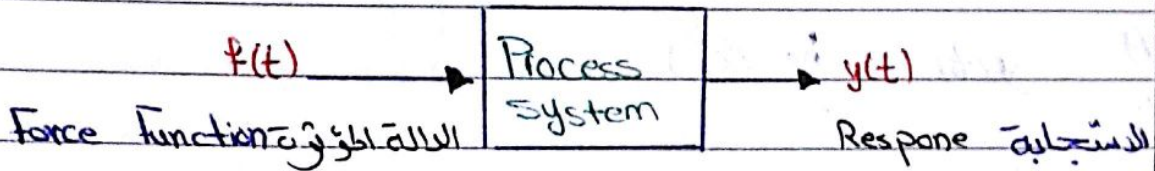


## Transfer Function

$$G(s) = \frac{y(s)}{F(s)} ; \frac{dy}{dt} \text{ State Variable}$$



**Transfer Function:** Is algebraic expression for the dynamic relation between the input "Force Function" and the Output of the process "Response" model and both expression is "S" domain  $y(s)$  and  $f(s)$ .

دالة جوية بين حالة دالة مدخلة تربط بين الدخل والمخرج

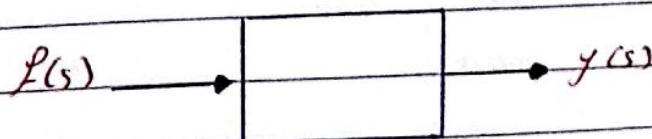
$$G(s) = \frac{\text{Output Variable in S domain}}{\text{Input Variable in S domain}}$$

$G(s) = K$  this mean Laplace transform of response Variable  
Laplace transform of Force Variable.

**Block diagram:** block diagram is graphical representation of their function and their interaction.

$$G(s) = \frac{(s+z_1)(s+z_2)+\dots+(s+z_n)}{(s+p_1)(s+p_2)+\dots+(s+p_n)}$$

Zero      الجذور  
Pole      الأقطاب



# ① Transfer Function of a first Order System.

$$a_1 \frac{dy}{dt} + a_0 y(t) = b_0 f(t)$$

$$\left(\frac{a_1}{a_0}\right) \frac{dy}{dt} + y(t) = \frac{b_0}{a_0} f(t)$$

$$\frac{a_1}{a_0} = T_p \quad ; \quad \frac{b_0}{a_0} = k_p$$

$$T_p \frac{dy}{dt} + y(t) = k_p f(t) \quad / \quad \text{Model of First Order System in time domain}$$

\* Assuming Zero Initial Condition; Taking laplace transform

$$T_p [s y(s) - y(0)] + y(s) = k_p f(s)$$

$$T_p s y(s) + y(s) = k_p f(s)$$

$$y(s) [T_p s + 1] = k_p f(s)$$

$$G(s) = \frac{y(s)}{f(s)} = \frac{k_p f(s)}{(T_p s + 1) f(s)} = \frac{k_p}{T_p s + 1} \quad / \quad \text{Transfer Function of}$$

First Order in laplace.

$T_p$  = Time Constant Process "Time".

$k_p$  = Steady state gain.

Unit Response Variable - نسبة الحالة المستقرة

Unit Force Variable

Example  $\left. \begin{array}{l} 50 \xrightarrow{+5} 55 \\ 30 \xrightarrow{+5} 35 \end{array} \right\} \frac{5}{5} = 1$

$\left. \begin{array}{l} 50 \xrightarrow{-5} 45 \\ 25 \xrightarrow{-2} 23 \end{array} \right\} \frac{-5}{-2} = \frac{5}{2}$

$k_p$  : give that ;  $+ve$  ,  $-ve$  ,  $>1$   
 $<1$

at steady state ( $t_p = 0$ ) ; ( $G(s) = k_p$ )

## ② Transfer Function For Second Order System.

$$a_2 \frac{d^2 y}{dt^2} + a_1 \frac{dy}{dt} + a_0 y(t) = b_0 f(t)$$

$$\left( \frac{a_2}{a_0} \right) \frac{d^2 y}{dt^2} + \left( \frac{a_1}{a_0} \right) \frac{dy}{dt} + y(t) = \left( \frac{b_0}{a_0} \right) f(t)$$

$$\text{let } T_p^2 = \frac{a_2}{a_0} ; T_p \eta = \frac{a_1}{a_0} ; k_p = \frac{b_0}{a_0}$$

$$\left[ T_p^2 \frac{d^2 y}{dt^2} + \eta T_p \frac{dy}{dt} + y(t) = k_p f(t) \right] \text{ Model of Second Order System "time"}$$

Assuming for Initial Condition ; Taking laplace transform

$$T_p^2 [s^2 y(s) - sy(0) - y'(0)] + \eta T_p [s y(s) - y(0)] + y(s) = k_p f(s)$$

$$T_p^2 S^2 y(s) + 2\zeta T_p S y(s) + y(s) = K_p f(s)$$

$$y(s) [T_p^2 S^2 + 2\zeta T_p S + 1] = K_p f(s)$$

$$G(s) = \frac{y(s)}{f(s)} = \frac{K_p}{T_p^2 S^2 + 2\zeta T_p S + 1} \quad \text{Transfer Function of Second Order.}$$

$T_p$  = Time Constant "Time".

$K_p$  = steady state gain.

$\zeta$  = Damping Factor.

### ③ Transfer Function of Zero Order System

$$a_0 y(t) = b_0 f(t)$$

$$a_0 y(s) = b_0 f(s)$$

$$G(s) = \frac{y(s)}{f(s)} = \frac{b_0}{a_0} = K_p \quad \text{Transfer Function of Zero Order system.}$$

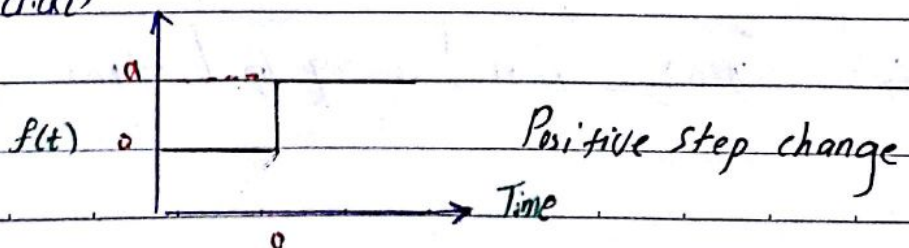
### ⊛ Force Function

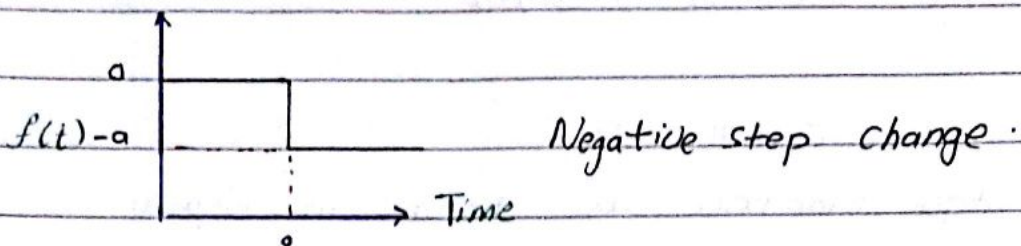
#### ① Step Function

$$f(t) = \begin{cases} 0 & t < 0 \\ a & t \geq 0 \end{cases}$$

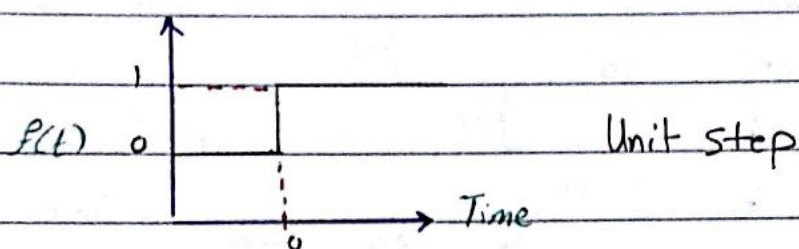
$$f(t) = a ; f(t) = a \cdot u(t)$$

$$F(s) = U(s) ; \frac{a}{s}$$





$$f(t) = \begin{cases} 0 & t < 0 \\ -a & t \geq 0 \end{cases} ; f(t) = -a \cdot u(t) ; f(s) = u(s) = \frac{-a}{s}$$

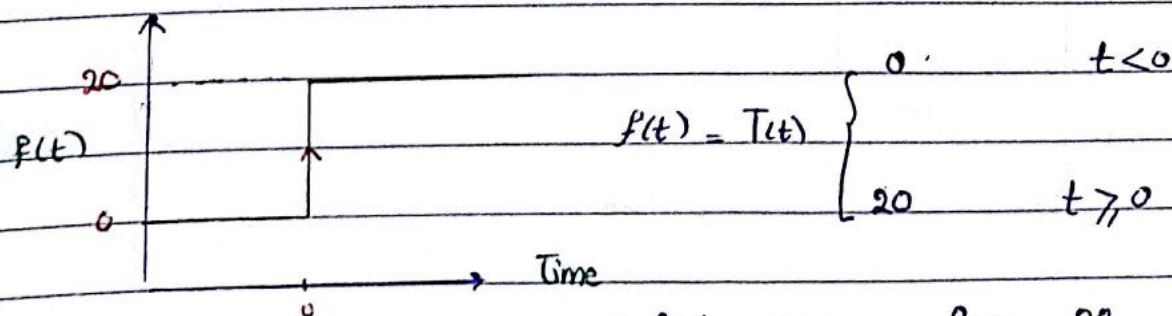
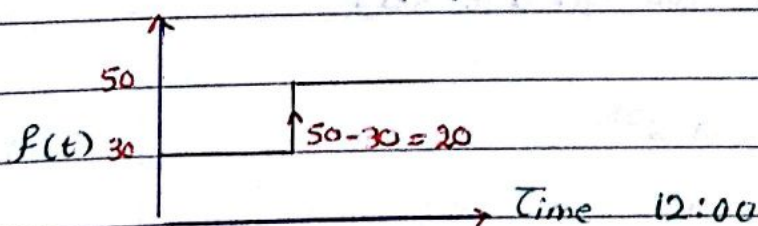


$$f(t) = u(t) = 1 ; f(s) = u(s) = \frac{1}{s}$$

Example:

$$T_1 = 30^\circ\text{C} ; T_2 = 50^\circ\text{C}$$

time = 12:00

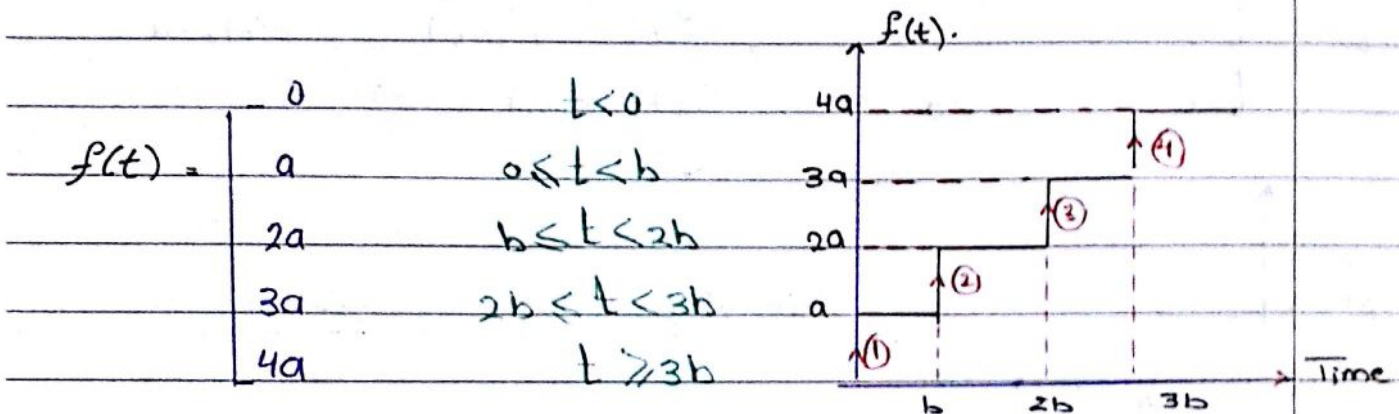


$$\therefore f(t) = 20 ; f(s) = \frac{20}{s}$$

$a =$  Magnitude of step function ; height.

## ② Stair Function دالة الدرج

This is ascires step Function as shown in Figure and Can be express as:

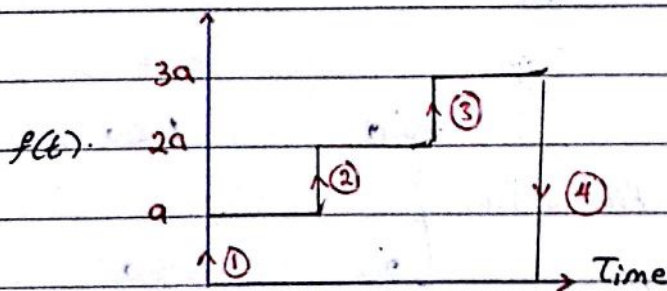


$$f(t) = a \cdot u(t) + a \cdot u(t-b) + a \cdot u(t-2b) + a \cdot u(t-3b).$$

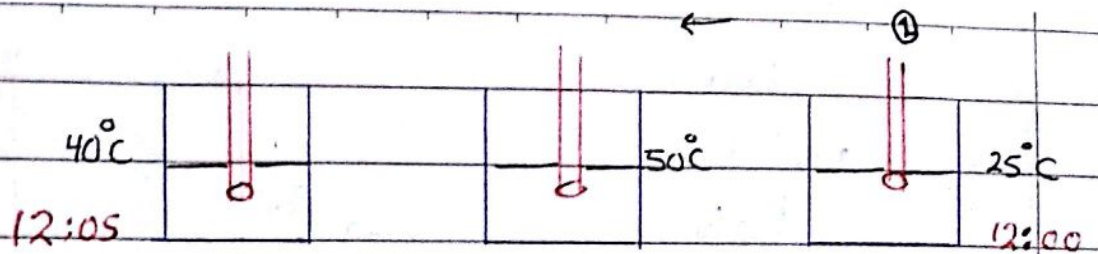
$$F(s) = \frac{a}{s} + \frac{a}{s} e^{-bs} + \frac{a}{s} e^{-2b} + \frac{a}{s} e^{-3bs}$$

$$f(t) = a \cdot u(t) + a \cdot u(t-b) + a \cdot u(t-2b) - 3a \cdot u(t-3b)$$

$$F(s) = \frac{a}{s} + \frac{a}{s} e^{-bs} + \frac{a}{s} e^{-2b} - 3 \frac{a}{s} e^{-3bs}.$$



Example 8



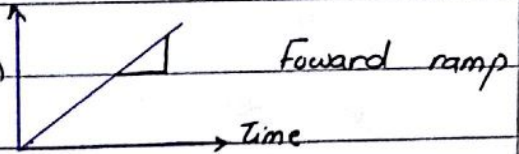
$$F(s) = \frac{25}{s} - \frac{10}{s} e^{-5s} ; f(t) = 25 - 10u(t-5)$$

### ③ Ramp Function

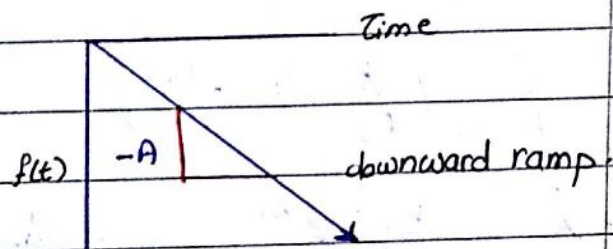
دالة المنحدر

The ramp function is a Uniform change with time.

$$f(t) = At ; f(s) = \frac{A}{s^2}$$



$$f(t) = -At ; f(s) = \frac{-A}{s^2}$$

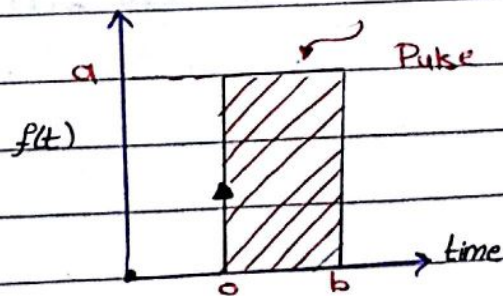


### ④ Rectangular pulse النبضة المستطيلة

Is positive step function change at period of the time and then negative step change.

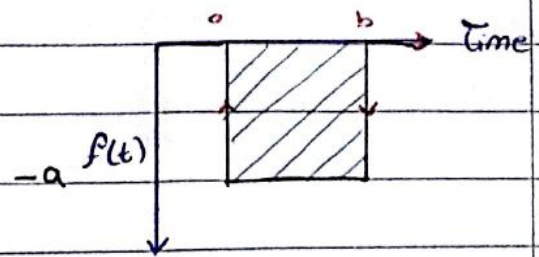
$$f(t) = \begin{cases} 0 & t < 0 \\ a & 0 \leq t \leq b \\ 0 & t > b \end{cases}$$

$$f(t) = a \cdot u(t) - a u(t-b)$$



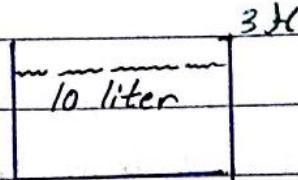
$$F(s) = \frac{a}{s} - \frac{a}{s} e^{-bs}$$

$$f(t) = \begin{cases} 0 & t < 0 \\ -a & a \leq t \leq b \\ 0 & t \geq b \end{cases}$$



⑤ Impulse " $\delta(t)$ " النبضة الانسية

1 liter = Area of Impulse



$$f(t) = \frac{K}{b} u(t) - \frac{K}{b} u(t-b)$$

$$f(s) = \frac{K}{bs} - \frac{K}{b} \frac{e^{-bs}}{s}$$

$$\mathcal{L}\{\delta(t)\} = \lim_{b \rightarrow 0} \left( \frac{K}{bs} (1 - e^{-bs}) \right)$$

$$= \lim_{b \rightarrow 0} \frac{K \cdot b \cdot e^{-bs}}{b} \text{ by L'Hopital Rule}$$

$$\mathcal{L}\{\delta(t)\} = K$$

	$f(t)$	$f(s)$
Unit step	1	$1/s$
Unit Impulse	$\delta(t)$	1

Impulse Function (ans)

**Example:** A System have the Following transfer Function

$$G(s) = \frac{O_o(s)}{O_i(s)} = \frac{3}{s} \quad \begin{matrix} \text{Out put} \\ \text{Input} \end{matrix}$$

a/ Find the differential equation?

b/ Find the response in time domain and sketch it for the following input Functions:

① Exponential Function  $\frac{1}{s} e^{-t/5}$  start after  $t=10$ ?

② Cosine Function start at  $t=4$ ,  $a=4$  and Frequency time of  $(\pi)$ .

Sol:

$$a/ \frac{O_o(s)}{O_i(s)} = \frac{3}{s}$$

$$s O_o(s) = 3 O_i(s) \Rightarrow \frac{1}{3} s O_o(s) = O_i(s) \quad \left( \frac{1}{3} \right) \text{ نضرب في } O_i \text{ لنبقى مع } O_o$$

$$\frac{1}{3} \frac{dO_o(t)}{dt} = O_i(t) \leftarrow \frac{1}{3} s O_o(s) - O_o(0) = 0 \quad \text{في حالة التحويل إلى} \quad \text{Transfer Function}$$

$$b/ G(s) = \frac{O_o(s)}{O_i(s)} = \frac{3}{s}; \quad O_i(t) = \frac{1}{s} e^{-t/5}$$

$$O_o(s) = \left( \frac{1}{s(s+1)} \right) e^{-10s} \quad \begin{matrix} O_o(s) = G(s) \cdot O_i(s) \\ O_o(t) = \mathcal{L}^{-1} G(s) \cdot O_i(s) \end{matrix}$$

$$O_o(t) = \mathcal{L}^{-1} O_o(s) = \mathcal{L}^{-1} \frac{3}{s} \left( \frac{1}{s(s+1)} \right) e^{-10s}$$

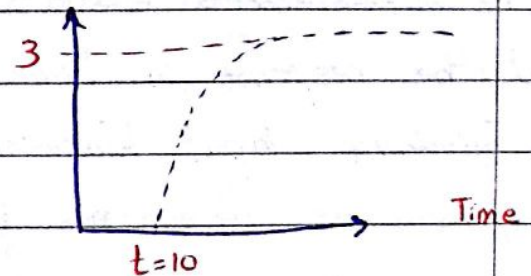
$$\frac{3}{s} \mathcal{L}^{-1} \left( \frac{1}{s(s+1)} \right) e^{-10s}$$

$$= \frac{3}{5} \int_{-1}^1 \frac{1}{s(s + \frac{1}{5})} e^{-xs} = \frac{3}{5} \int_{-1}^1 \frac{5}{s(5s+1)} e^{-xs}$$

$$= \frac{3}{5} \left[ \frac{1}{\frac{1}{5}} \left( 1 - e^{-\frac{1}{5}(t-10)} \right) \right]$$

$$Q_o(t) = 3 \left( 1 - e^{-(t-10)/5} \right)$$

t	$Q_{out}$
10	-
11	-
12	-



b/2  $Q_i(t) = a \cos \omega(t-t_d)$

$$\omega = \frac{2\pi}{\text{time}} = \frac{2\pi}{\pi} = 2$$

$$\omega = 2\pi f$$

$$Q_i(t) = 4 \cos 2(t-4)$$

Time

$Q_{out}$

$$Q_i(t) = 4 \cos(2t - 8)$$

4

-

5

-

$$Q_i(s) = \frac{4s}{s^2 + 4} e^{-4s}$$

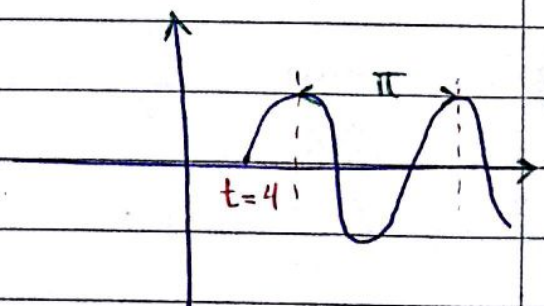
6

-

$$Q_o(t) = \mathcal{L}^{-1} \left( \frac{3}{s} \right) \left( \frac{4s}{s^2 + 4} \right) e^{-4s}$$

$$Q_o(t) = 3 \sin 2(t-4)$$

$$Q_o(t) = 3 \sin(2t - 8)$$



Example: Continuous stirred tank heater ;

متغير الزمن

Heat Balance :

S.S (steady state)  $\ominus$

$$\bar{Q} + m c_p \bar{T}_i - m c_p \bar{T} = \rho c_p V \frac{dT}{dt} = 0 \quad (1)$$

Unsteady state (t)

In-Out = Accumulation

$$\dot{Q}(t) + m c_p \dot{T}_i(t) - m c_p \dot{T}(t) = \rho V c_p \frac{dT(t)}{dt} \quad (2)$$

$$\frac{kJ}{kg \cdot ^\circ C} + \frac{kg}{s} \cdot \frac{kJ}{kg \cdot ^\circ C} \cdot e = \frac{kg}{m^3} \cdot \frac{m^3}{s} \cdot \frac{kJ}{kg \cdot ^\circ C} \cdot e$$

2-1 gives

لدينا يكون ثابت الحرارة

$$m c_p (\dot{T}_i(t) - \bar{T}_i) + (\dot{Q}(t) - \bar{Q}) - m c_p (\dot{T}(t) - \bar{T}) =$$

$$\rho V c_p \frac{d}{dt} (T(t) - \bar{T})$$

$$m c_p \dot{T}_i(t) + \dot{Q}(t) - m c_p \dot{T}(t) = \rho V c_p \frac{dT(t)}{dt}$$

k = Input - Output  
m c\_p

$$\dot{T}_i(t) + \frac{\dot{Q}(t)}{m c_p} - \dot{T}(t) = \left( \frac{\rho V}{m} \right) \frac{dT(t)}{dt}$$

$$\tau \frac{dT}{dt} + T(t) = \dot{T}_i(t) + \frac{\dot{Q}(t)}{m c_p k}$$

$$\tau = \frac{\rho V}{m}$$

$$k = \frac{1}{m c_p}$$

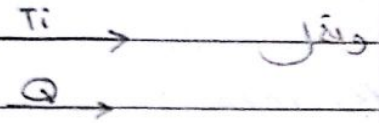
$$T_s T(s) + T(s) = \dot{T}_i(s) + k \dot{Q}(s)$$

$$T(s) [T_s + 1] = \dot{T}_i(s) + k \dot{Q}(s)$$

$$T(s) = \frac{1}{T_s + 1} \dot{T}_i(s) + \frac{k}{T_s + 1} \dot{Q}(s)$$

Now: if  $Q$  is constant  $Q(s) = 0$  input  $T_i$  is initial value

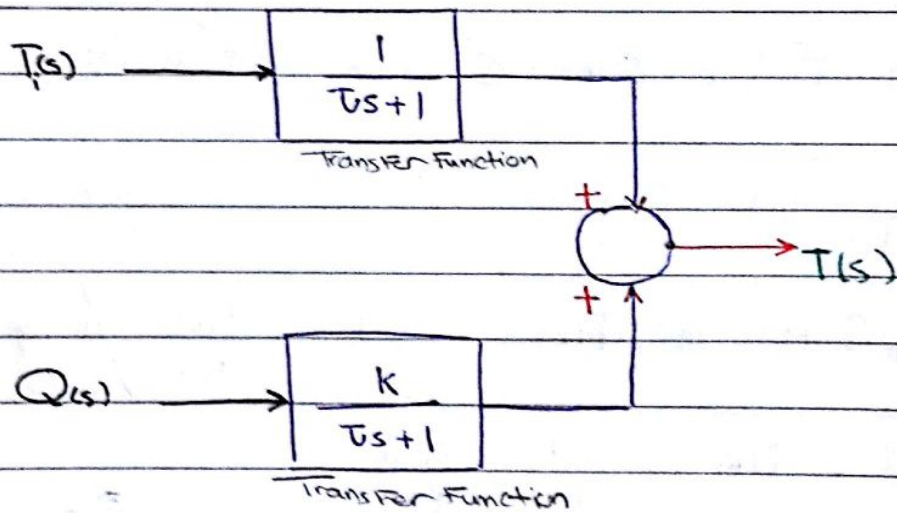
$$\frac{T(s)}{T_i(s)} = \frac{1}{Ts+1} = G(s)$$



If  $T_i$  is constant  $T_i(s) \rightarrow 0$

$$\frac{T(s)}{Q(s)} = \frac{k}{Ts+1} = G(s)$$

Block diagram



\* معادلة لا خطية الاصل  $\neq 1$

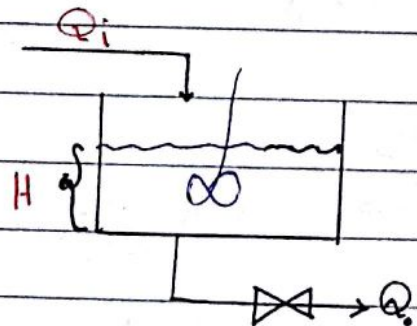
تحويل الدالة اللاخطية إلى خطية \* لا يمكن تحويل SS و Un الا إذا كانت خطية

عان الطريقة القياسية لتحويل الدالة اللاخطية إلى خطية وذلك باستخدام مفكول Taylor

$$f(x) = f(x_0) + \frac{df}{dx} \bigg|_{x_0} (x - x_0) + \frac{1}{2!} \frac{d^2f}{dx^2} \bigg|_{x_0} (x - x_0)^2$$

$$Q_0 = k \sqrt{H}$$

معادلة لا خطية



$$Q = k \sqrt{H_0} + \frac{k}{2\sqrt{H_0}} (H - H_0)$$

Example: Mass Balance

استجابة الخطوة Response:

$$PQ_i - PQ_0 = PA \frac{dH}{dt} = 0 \quad \text{SS} \quad \text{①}$$

$$PQ_i(t) - PQ_0(t) = PA \frac{dH(t)}{dt} \quad \text{Un SS} \quad \text{in - Out = Accumulation}$$

$$\underline{Q_0(t) = k H^{1/2}} \quad \text{لا خطية Un SS} \quad \text{At Steady state} \quad \underline{Q_0 = k \sqrt{H}}$$

↓ Unsteady state

$$Q_0(t) = k \bar{H}^{1/2} + \frac{k}{2\bar{H}^{1/2}} (H(t) - \bar{H})$$

$$PQ_i(t) - Pk\bar{H}^{1/2} + \frac{k}{2\bar{H}^{1/2}} (H(t) - \bar{H}) = PA \frac{dH(t)}{dt} \quad \text{②}$$

معادلة خطية

Subtract

$$\underline{Q_0(t)}_{\text{Un SS}} - \underline{Q_0}_{\text{S.S}} = \left[ k\bar{H}^{1/2} + \frac{k}{2\bar{H}^{1/2}} H(t) - k\bar{H}^{1/2} \right] = A \frac{d(H(t) - \bar{H})}{dt} \Rightarrow 2-1 \text{ مرتبة}$$

$$Q_i(t) - \frac{k}{2H^{1/2}} H(t) = A \frac{dH(t)}{dt}$$

$$A \frac{dH(t)}{dt} + \frac{k}{2H^{1/2}} H(t) = Q_i(t)$$

$$A (sH(s) - H(0)) + \frac{k}{2H^{1/2}} H(s) = Q_i(s)$$

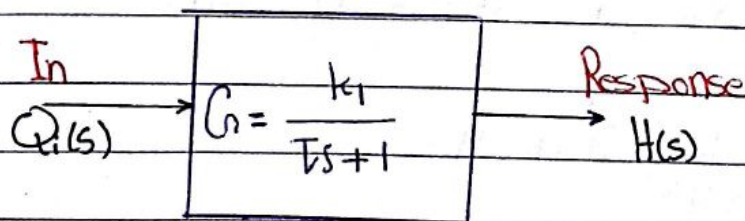
$$H(s) \left( As + \frac{k}{2H^{1/2}} \right) = Q_i(s)$$

$$G(s) = \frac{H(s)}{Q_i(s)} = \frac{1}{As + \frac{k}{2\sqrt{H}}} = \left( \frac{2\sqrt{H}/k}{\frac{A2\sqrt{H}}{k}s + 1} \right)$$

$$\text{Or } G(s) = \frac{k_1}{Ts + 1} ; k_1 = \frac{2\sqrt{H}}{k}$$

$$T = \frac{2A\sqrt{H}}{k}$$

Block diagram



H.W/ For the above example if the following data are given  $A = 1 \text{ m}^2$ ,  $\bar{H} = 1.44 \text{ m}$ ,  $\bar{Q}_i = \bar{Q}_o = 2.4 \times 10^{-3} \text{ m}^3/\text{s}$  and a step change in Inlet Flow rate =  $0.6 \times 10^{-3} \text{ m}^3/\text{s}$  Find the response and sketch it

ans/  $\hat{H}(t) = 2.16 \text{ m}$  (Final Value).

تحويل الدالة اللاخطية التي تعتمد على متغيرين أو أكثر إلى خطية

$$f(x, y) = f(x_0, y_0) + \left. \frac{df}{dx} \right|_{x_0, y_0} (x - x_0) + \left. \frac{df}{dy} \right|_{x_0, y_0} (y - y_0)$$

Example: Change the Following Function to linear

$$f(x, y) = x \cdot y \quad y_0 = 5, \quad x_0 = 2$$

$$\therefore f(x, y) = x_0 \cdot y_0 + y_0 (x - x_0) + x_0 (y - y_0)$$

$$= x_0 \cdot y_0 + y_0 x - y_0 x_0 + x_0 y - x_0 y_0$$

$$f(x, y) = 5x + 2y - 10$$

**Example:** A Well-stirred tank Contain  $(M)$  kg of a liquid with Specific heat Capacity of  $(J/kg \cdot ^\circ C)$ . The liquid is heated electrically with Constant rate of  $(Q)$ . At Steady state the liquid inlet at a rate of  $m$  kg/h a temp of  $T_i$  and Outlet at  $T_o$ . Find the response to step change in inlet Flow rate of 20% /h ? From steady state ?

Sol / S. State.

.. In Step change

في  $m(t)cpT_o$  نثبت مقدار وتغير الآخر

$$\bar{Q} + \bar{m}cp\bar{T}_i - mcp\bar{T}_o = \mu cp \frac{d\bar{T}_o}{dt} = 0 \quad (1)$$

نثبت  $(m)$  وذلك باعتبار  $Q$  ذلك

Unsteady state

Taylor Series at (3).

$$\bar{Q} + \underline{\dot{m}(t)}cp\bar{T}_i - \dot{m}(t)cp\bar{T}_o(t) = \mu cp \frac{d\bar{T}_o(t)}{dt} \quad (2)$$

$$\dot{m}(t)\bar{T}_o(t) = \bar{m}\bar{T}_o + \bar{T}_o(\dot{m}(t) - \bar{m}) + \bar{m}(\bar{T}_o(t) - \bar{T}_o) \quad (3)$$

وجود تغير في الحرارة  
الحق اليقين مع ذلك بالآخر

$$\therefore \dot{m}(t)cp\bar{T}_i - \bar{m}cp\bar{T}_o - (\bar{m}cp\bar{T}_o + cp\bar{T}_o(\dot{m}(t) - \bar{m}) +$$

$$2-1 \text{ و } \bar{m}cp\bar{T}_o(t) - \bar{T}_o) + \bar{m}cp\bar{T}_o = \mu cp \frac{d\bar{T}_o(t)}{dt}$$

$$m(t)cp\bar{T}_i - cp\bar{T}_o m(t) - cp\bar{m}\bar{T}_o(t) = \mu cp \frac{d\bar{T}_o(t)}{dt}$$

$$(cp\bar{T}_i - cp\bar{T}_o)m(t) - cp\bar{m}\bar{T}_o(t) = \mu cp \frac{d\bar{T}_o(t)}{dt}$$

$$\left( \frac{\bar{\theta}_i - \bar{\theta}_o}{\mu} \right) \bar{m}(t) - \frac{\bar{m}}{\mu} \bar{\theta}_o(t) = \frac{d\bar{\theta}_o(t)}{dt}$$

$$\frac{\mu}{\bar{m}} \frac{d\bar{\theta}_o(t)}{dt} + \bar{\theta}_o(t) = \left( \frac{\bar{\theta}_i - \bar{\theta}_o}{\bar{m}} \right) \bar{m}(t)$$

$$(2) \tau \frac{d\bar{\theta}_o(t)}{dt} + \bar{\theta}_o(t) = k \bar{m}(t)$$

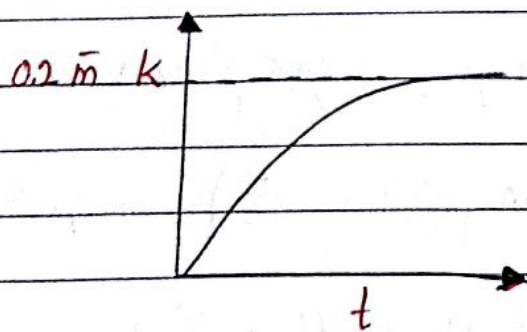
$$\tau s \bar{\theta}_o(s) + \bar{\theta}_o(s) = k \bar{m}(s)$$

$$\bar{\theta}_o(s) [\tau s + 1] = k \bar{m}(s) \quad k = \frac{\bar{\theta}_i - \bar{\theta}_o}{\bar{m}}$$

$$\bar{\theta}_o(s) = \frac{\bar{\theta}_o(s)}{\bar{m}(s)} = \frac{k}{\tau s + 1} \quad \tau = \frac{\mu}{\bar{m}}$$

$$\bar{\theta}_o(s) = \frac{k}{\tau s + 1} \bar{m}(s) \quad ; \quad \bar{m}(s) = \frac{0.2 \bar{m}}{s} \text{ Steady state.}$$

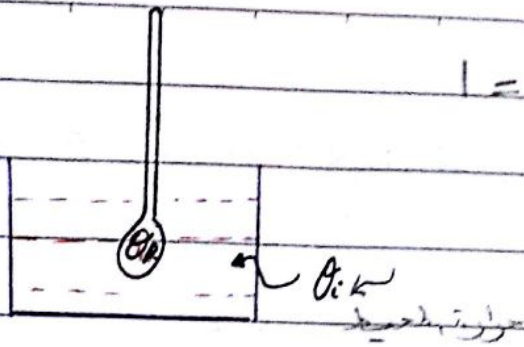
$$\bar{\theta}_o(s) = 0.2 \bar{m} \left( \frac{k}{s(\tau s + 1)} \right) = 0.2 \bar{m} k \left( 1 - e^{-st/\tau} \right)$$



# First Order System"

المثال 5 = 1

Heat Balance:



$$In = Out + Acc$$

$$In = Acc$$

المجموع = 0

at steady state

$$hA(\overline{\theta_i} - \overline{\theta_R}) = Mcp \frac{d\overline{\theta_R}}{dt} = 0 \quad (1)$$

Unsteady state

$$hA(\theta_i - \theta_R) = Mcp \frac{d\theta_R}{dt} \quad (2)$$

Subtract 2-1

$$hA(\theta_i - \theta_R - \overline{\theta_i} + \overline{\theta_R}) = Mcp \frac{d(\theta_R' - \overline{\theta_R})}{dt}$$

$$hA(\theta_i(t) - \theta_R(t)) = Mcp \frac{d\theta_R(t)}{dt}$$

$$\theta_i(t) - \theta_R(t) = \frac{Mcp}{hA} \frac{d\theta_R(t)}{dt}$$

المعدل عند التسخين = 1

$$\text{let } \frac{Mcp}{hA} = \tau$$

$$\tau \frac{d\theta_R(t)}{dt} - \theta_i(t) = \theta_R(t)$$

Taking Laplace Transform

T.F  $\theta_R(s) = 0$  شرط

$$\tau (s\theta_R(s) - \theta_R(0)) - \theta_i(s) = \theta_R(s) \quad k \text{ For Thermometer} = 1$$

$$(\tau s + 1) \theta_R(s) = \theta_i(s)$$

$$\therefore G(s) = \frac{\theta_R(s)}{\theta_i(s)} = \frac{1}{\tau s + 1}$$

\* Transform Function of 1<sup>st</sup> Order System

$$G(s) = \frac{y(s)}{x(s)} = \frac{k}{\tau s + 1} \quad \leftarrow \text{General Form}$$

$k$  : steady state gain

$\tau$  : Time Constant

Notice that steady state gain for the thermometer = 1

20. إذا تم تصنيف أجهزة First Order System

\*

لها ثابت زمني  $(\tau)$  = 1

أولاً :

التحليل الجبري لنظام دالة التحويل من الدرجة الأولى  $(\tau s + 1)$

ثانياً :

نسب الحالة المستقرة  $K =$  وحالة خاصة للحرارة فإن

ثالثاً :

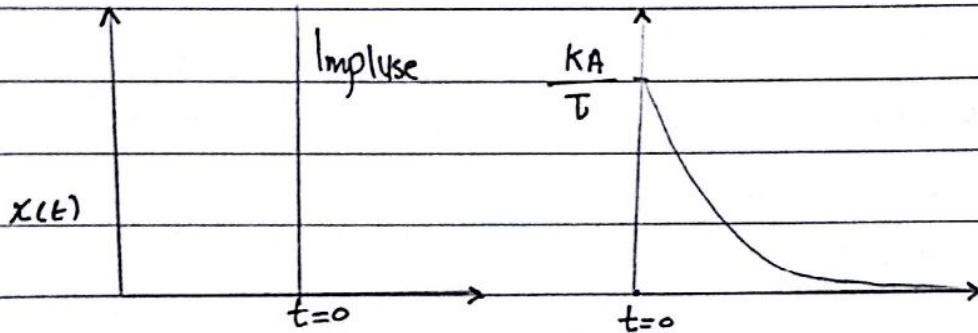
$$K = 1$$

1

## Response of 1<sup>st</sup> Order element to Impulse force

$$G(s) = \frac{y(s)}{x(s)} = \frac{k}{\tau s + 1} \quad ; \quad x(s) = A \quad \text{Impulse}$$

$$y(s) = kA \left[ \frac{1}{\tau s + 1} \right] \Rightarrow y(t) = \frac{kA}{\tau} e^{-t/\tau}$$

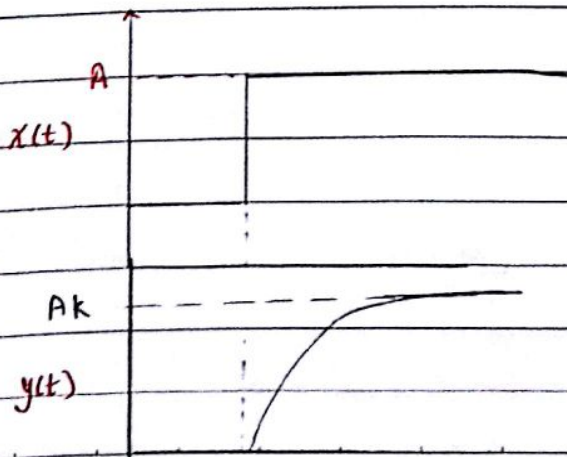


## Response of 1<sup>st</sup> Order System to step force.

$$G(s) = \frac{y(s)}{x(s)} = \frac{k}{\tau s + 1} \quad x(s) = \frac{A}{s}$$

$$y(s) = kA \left( \frac{1}{s(\tau s + 1)} \right) \quad ; \quad \frac{A}{s} \left[ \frac{k}{\tau s + 1} \right]$$

$$y(t) = kA \left( 1 - e^{-t/\tau} \right)$$



Response of 1<sup>st</sup> Order to pulse change. step<sup>+</sup>  
step<sup>-</sup>

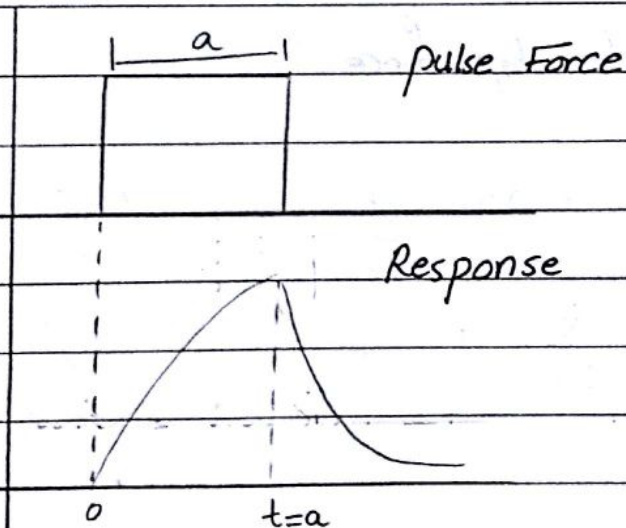
$$G(s) = \frac{k}{\tau s + 1} = \frac{y(s)}{x(s)}, \quad x(s) = \frac{A}{s} (1 - e^{-as})$$

$$y(s) = \frac{kA}{s} \frac{(1 - e^{-as})}{(\tau s + 1)}$$

$$y(s) = \frac{kA}{s(\tau s + 1)} - \frac{kA e^{-as}}{s(\tau s + 1)}$$

$$y(s) = \frac{kA}{s(\tau s + 1)} - \frac{kA e^{-as}}{s(\tau s + 1)}$$

$$y(t) = kA(1 - e^{-t/\tau}) - kA(1 - e^{-\frac{(t-a)}{\tau}})$$



# Response of 1<sup>st</sup> Order System to Sine Function Force

$$G(s) = \frac{P_o(s)}{P_i(s)} = \frac{k}{Ts + 1}$$

$$P_i(t) = a \sin \omega t \quad ; \quad P_i(s) = \frac{a\omega}{s^2 + \omega^2}$$

$$P_o(s) = \left( \frac{k}{Ts + 1} \right) \left( \frac{a\omega}{s^2 + \omega^2} \right)$$

$$P_o(s) = k a \omega \left( \frac{1}{(Ts + 1)(s^2 + \omega^2)} \right) \xrightarrow{\text{استخدام التجزئة}} \frac{A}{Ts + 1} + \frac{B}{s^2 + \omega^2}$$

$$P_o(t) = \frac{a k \omega T}{T^2 \omega^2 + 1} e^{-t/T} + \frac{a k}{\sqrt{T^2 \omega^2 + 1}} \sin(\omega t + \phi)$$

↑ exponential      ↑ Oscillatory      ↑

$$\text{Or } P_o(t) = \frac{a k}{\sqrt{T^2 \omega^2 + 1}} \sin(\omega t + \phi) \quad t \rightarrow \infty$$

## Amplitude Ratio (AR)

نسبة الاستجابة  
سعة الدالة المؤثرة = نسبة السعة

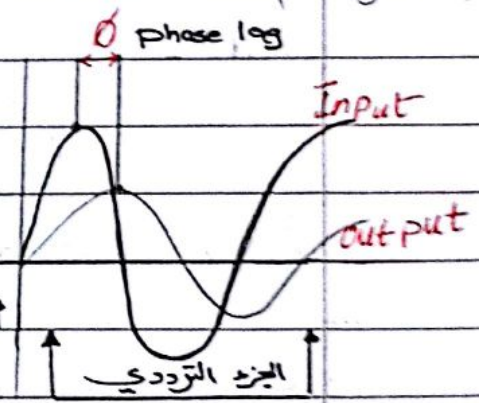
$$AR = \frac{a k}{\sqrt{T^2 \omega^2 + 1}} = \frac{k}{\sqrt{T^2 \omega^2 + 1}}$$

$\omega = 2\pi f$  ,  $f$  - frequency  
 $\omega$  - Angular frequency.

$\phi$  phase lag in radian

$$\text{frequency time} = \frac{1}{f}$$

$$t = \frac{|\phi|}{\omega} \text{ phase lag in time}$$



$$\phi = \frac{180}{\pi} \text{ phase lag in degree.}$$

**Example:** A thermometer have a Constant time of 6 sec is placed in a liquid of  $100^\circ\text{C}$  for a long time if the input force is a Sine wave of amplitude  $a = 2^\circ\text{C}$  and a frequency of  $\frac{1}{6\pi}$  Hz. Find the Response in time domain and sketch it.

**Sol/**  $\omega = 2\pi f = \frac{2\pi}{6\pi} = \frac{1}{3} \text{ rad/sec}; \phi = \tan^{-1}(-\omega\tau) = -1.11 \text{ rad}$

$\tau = 6 \text{ sec}, a = 2$

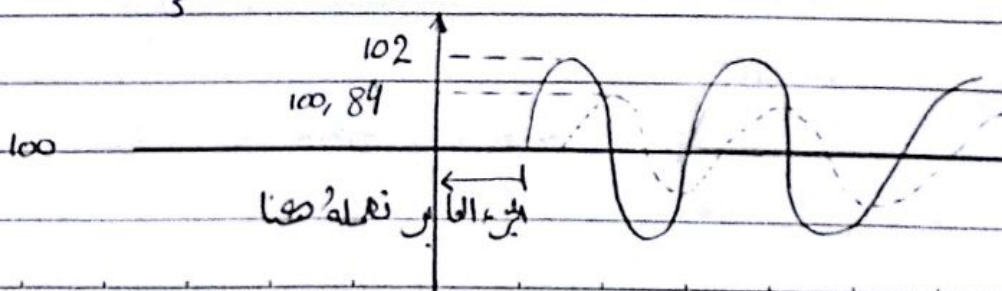
$$\theta_o(t) = \frac{a k}{\sqrt{\tau^2 \omega^2 + 1}} \sin(\omega t + \phi) \text{ or } \phi = \tan^{-1}(-\omega\tau) \text{ or } \phi = \tan^{-1}(\omega\tau)$$

$$\theta_o(t) = \frac{1}{\sqrt{6^2 \left(\frac{1}{3}\right)^2 + 1}} \sin\left(\frac{1}{3}t - 1.11\right)$$

$$\theta_o(t) = 0.89 \sin\left(\frac{1}{3}t - 1.11\right) \quad \text{Rad} \quad \text{التوقيت}$$

$$\phi = -1.11 \text{ rad}; \phi = \tan^{-1}(-\omega\tau) \text{ Phase lag}$$

$$\frac{|\phi|}{\omega} = \frac{1.11}{\frac{1}{3}} = 3.33 \text{ Sec in time}$$



Example: A Sine wave of frequency ( $\frac{1}{\pi}$  Hz) and amplitude of 5 is affect On system whose T.F is

$$G(s) = \frac{y(s)}{x(s)} = \frac{1}{s+2} = \frac{k}{Ts+1}$$

معادلة مرتبة أولى للبركان تكون هكذا

Show that  $y(t) = \frac{5}{4} e^{-2t} + \frac{5}{2\sqrt{2}} \sin(114^\circ t - 45^\circ)$

الجزء الأيمن أول سويتا

Sol

$$G(s) = \frac{1}{s+2} = \frac{\frac{1}{2}}{\frac{1}{2}s+1} = \frac{k}{Ts+1} \quad k = \frac{1}{2} ; T = \frac{1}{2}$$

$$y(t) = \frac{kaw\tau}{\tau^2\omega^2+1} e^{-t/\tau} + \frac{ak}{\sqrt{\tau^2\omega^2+1}} \sin(\omega t + \phi)$$

$$\phi = \tan^{-1}(-\omega\tau) \quad , \quad \omega = 2 \quad , \quad a = 5$$

$$y(t) = \frac{5(\frac{1}{2})(2)(\frac{1}{2})e^{-t/\frac{1}{2}}}{(\frac{1}{2}+2)+1} + \frac{5 \times \frac{1}{2}}{\sqrt{(\frac{1}{2}+2)^2+1}} \sin(2t + \tan^{-1}(2 \times \frac{1}{2}))$$

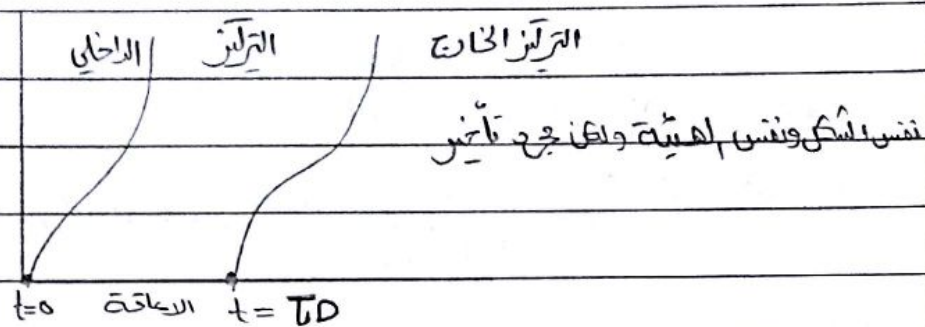
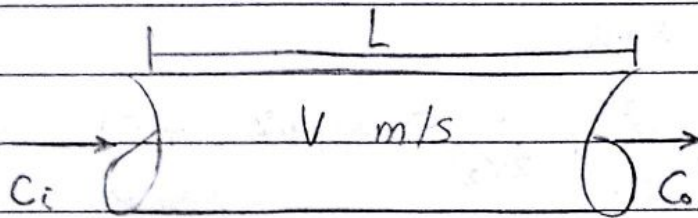
$$y(t) = \frac{5}{4} e^{-2t} + \frac{5}{2\sqrt{2}} \sin(2t - 0.7854)$$

$$y(t) = \frac{5}{4} e^{-2t} + \frac{5}{2\sqrt{2}} \sin(114^\circ t - 45^\circ)$$

## Pure time delay

أجهزة القَوَّع الزمني للفحص

هي تلك الأجهزة التي لا يطرأ فيها أي تغير على مقدار أو شكل الإشارة الداخلة بل يكون تأثير الجهاز عن إعاقة زمنية محض



## T.F of pure time delay

$$G(s) = \frac{y(s)}{f(s)} = e^{-T_D s} \rightarrow \text{الإعاقة والتأخير } T_D$$

$$T_D = \frac{L}{V}$$

**Example:** An insulated tube transfer a liquid with velocity of  $0.5 \text{ m/s}$ . At time  $= 0$ , it's temp is exposed to Sine change  $Q_i(t) = 10 \sin 2t$  find the response of a thermometer placed at a distance  $20 \text{ m}$  away from the Inlet?

$$G(s) = \frac{Q_o(s)}{Q_i(s)} = e^{-T_D s}$$

$$Q_o(s) = Q_i(s) e^{-\tau DS}$$

$$\frac{Q_o(s)}{Q_i(s)} = e^{-\tau DS}$$

$$Q_o(t) = Q_i(s) e^{-\tau DS}$$

$$Q_o(t) = \mathcal{L}^{-1} Q_i(s) e^{-\tau DS}$$

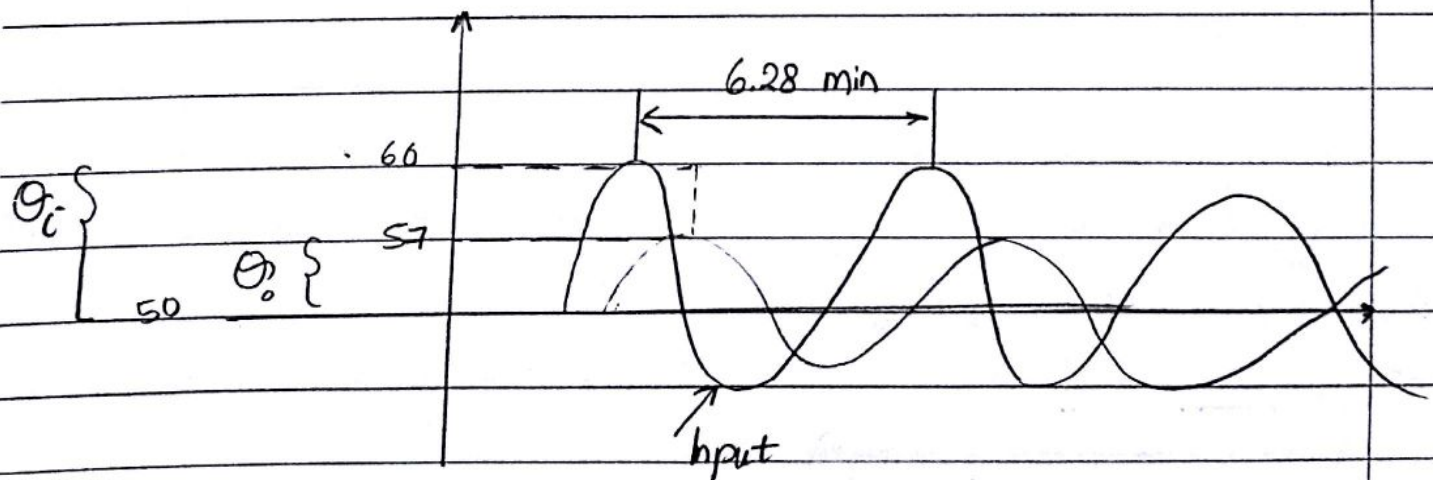
$$Q_o(t) = \mathcal{L}^{-1} \frac{a\omega}{s^2 + \omega^2} e^{-\tau DS}$$

$$\tau D = \frac{L}{V} = 40$$

$$Q_{out} = \mathcal{L}^{-1} \frac{10(2)}{s^2 + 4} e^{-40s}$$

$$Q_o(t) = 10 \sin 2(t - 40)$$

**H.w/** A first Order System is exposed to a Sinewave and the sketch belows show both the Input and the Output response. Find T.F of the System and the Input and Output Force?



كيفية تعيين ثابت زمن جهاز مرتبة أولى عملياً

يتم تعيين ثابت جهاز مرتبة أولى وذلك بتعريض الجهاز الى تغيّر درجتي تغيّر مفاجئ ومن خلال اخذ عدة قراءات لاستجابة مع الزمن نتمكن من تعيين ثابت زمن الجهاز وكذلك 8

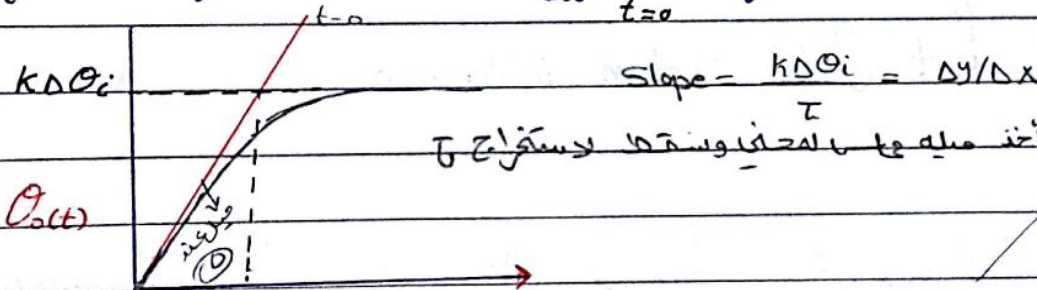
## ① Initial Slope

طريقة الميل الابتدائي

$$Q_o(s) = G(s) \cdot Q_{is}(s) \quad ; \quad Q_o(s) = \frac{k}{\tau s + 1} \cdot \frac{Q_{is}(s)}{s}$$

$$Q_o(t) = k \Delta \theta_i (1 - e^{-t/\tau}) \quad \text{Step change } \leftarrow \frac{1}{s(\tau s + 1)} \text{ 1st}$$

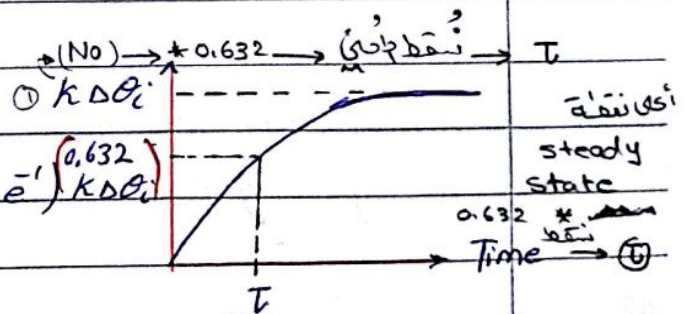
$$\frac{dQ_o(t)}{dt} = \frac{k \Delta \theta_i}{\tau} e^{-t/\tau} \quad ; \quad \frac{dQ_o(t)}{dt} \Big|_{t=0} = \frac{k \Delta \theta_i}{\tau}$$



## ② 0.632 from the response

$$Q_o(t) = k \Delta \theta_i (1 - e^{-t/\tau})$$

$$\text{if } t = \tau \rightarrow Q_o(t) = k \Delta \theta_i (1 - e^{-1}) = 0.632 k \Delta \theta_i$$



$$Q_o(t) = 0.632 k \Delta \theta_i$$

$$\text{at } t = \tau, \quad Q_o(t) = 0.632 k \Delta \theta_i$$

↓ steady state

③ طريقة الرسم المنفرد لوغاريتمي

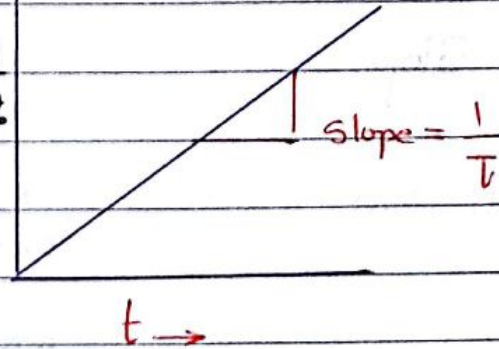
$$Q(t) = K \Delta \theta_i (1 - e^{-t/\tau})$$

$$= K \Delta \theta_i - K \Delta \theta_i e^{-t/\tau}$$

$$\frac{K \Delta \theta_i - Q(t)}{K \Delta \theta_i} = e^{-t/\tau}$$

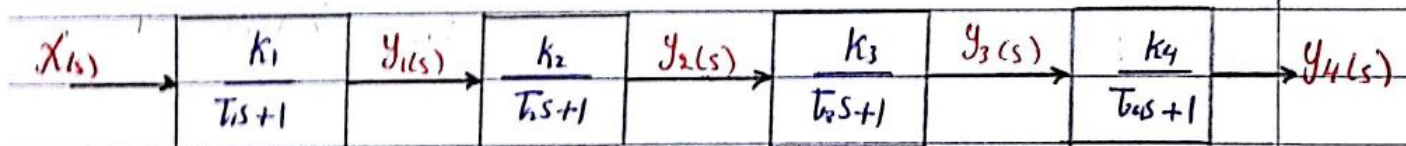
$$-\ln \frac{K \Delta \theta_i - Q(t)}{K \Delta \theta_i} = t/\tau$$

$$\ln \frac{K \Delta \theta_i}{K \Delta \theta_i - Q}$$



$$\ln \frac{K \Delta \theta_i}{K \Delta \theta_i - Q(t)} = t/\tau$$

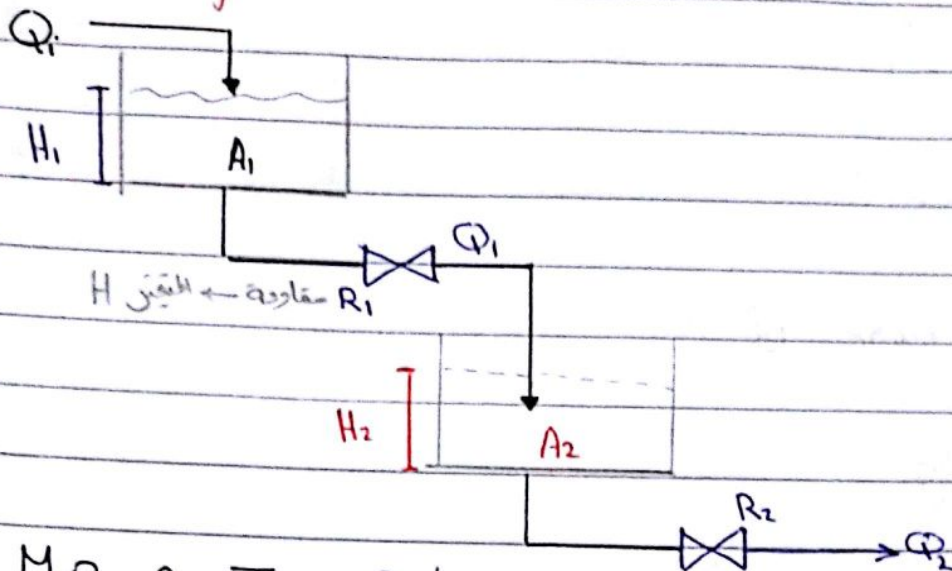
استجابة النظام من الدرجة الأولى في الخدمة



- 1- Non interacting System
- 2- Interacting System

Non-interacting system: أي دالة تحويل أي جهاز لا تحتوي على متغير التخميد السابقة

For Example



M.B On Tank ① In. Out = Acc.

$$AQ_i(t) - AQ_1(t) = A_1 \frac{dH_1(t)}{dt}$$

if the Flow is laminar (linear Relationship  $Q$  of  $H_1$ )

$$\left( Q_1 = \frac{H_1(t)}{R_1} \right) \Rightarrow Q_i(t) - \frac{H_1(t)}{R_1} = A_1 \frac{dH_1(t)}{dt} \quad \left( \text{Valve + R موجود} \right) \quad \left( Q_1 = \frac{H_1(t)}{R_1} \right) \quad \text{Flow laminar}$$

$$T_1 \frac{dH_1(t)}{dt} + H_1(t) = R_1 Q_i(t) \quad ; \quad T_1 = A_1 R_1$$

$$H_1(s) (T_1 s + 1) = R_1 Q_i(s)$$

$$\frac{H_1(s)}{Q_i(s)} = \frac{R_1}{T_1 s + 1} \quad \text{① Response Out}$$

if a step change occurs in the Inlet Flow Rate

$$Q_i(s) = \frac{\Delta Q_i}{s}$$

$$H_1(s) = \frac{\Delta Q_1}{s} \left( \frac{R_1}{\tau_1 s + 1} \right) \quad \text{Transfer Function}$$

$$H_1(t) = R_1 \Delta Q_1 (1 - e^{-t/\tau}) \quad \text{First Order Step Change}$$

gain & time

Mass Balance at tank ②

$$Q_1(t) - Q_2(t) = A_2 \frac{dH_2}{dt}$$

$$\tau_2 \frac{dH_2(t)}{dt} + H_2(t) = R_2 Q_1(t)$$

$$(\tau_2 s + 1)(H_2(s)) = R_2 Q_1(s)$$

$$\frac{H_2(s)}{Q_1(s)} = \frac{R_2}{\tau_2 s + 1} \quad \text{②}$$

but  $Q_1(t) = \frac{H_1(t)}{R_1}$

$$\frac{H_2(s)}{H_1(s)} = \frac{R_2}{R_1} \left( \frac{1}{\tau_2 s + 1} \right)$$

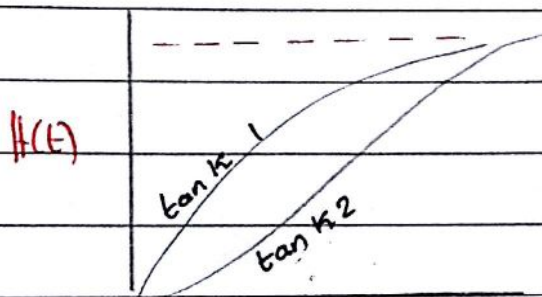
$$\frac{H_1(s)}{Q_1(s)} = R_1 \left( \frac{1}{\tau_1 s + 1} \right)$$

$$\frac{H_2(s)}{Q_1(s)} = \frac{R_2}{(\tau_1 s + 1)(\tau_2 s + 1)}$$

$$H_2(s) = \frac{R_2}{(T_1 s + 1)(T_2 s + 1)} Q_i(s)$$

if a step change occurs in inlet flow rate  $Q_i(s) = \frac{\Delta Q_i}{s}$

$$H_2(t) = \Delta Q_i R_2 \left( 1 - \frac{T_2 e^{-t/T_1} - T_1 e^{-t/T_2}}{T_1 - T_2} \right)$$



Block diagram



Now if  $A_1 = A_2$  &  $R_1 = R_2 = R$ ,  $T_1 = T_2 = T$

$$H_2(s) = \frac{R_2}{(T s + 1)^2} Q_i(s) \quad \text{if} \quad Q_i(s) = \frac{\Delta Q_i}{s}$$

$$H_2(t) = \int_0^t \frac{\Delta Q_i R_2}{s (T s + 1)^2} dt$$

$$h(t) = \Delta Q \cdot R_2 \left( 1 - e^{-t/\tau} - \frac{t}{\tau} e^{-t/\tau} \right)$$

In general

$$\frac{Y(s)}{X(s)} = \frac{k_1 k_2 k_3}{(\tau_1 s + 1)(\tau_2 s + 1)(\tau_3 s + 1) \dots}$$

if  $k = k_1 = k_2 = k_3$   
 $\tau = \tau_1 = \tau_2 = \tau_3 \dots$

$$\frac{Y(s)}{X(s)} = \frac{k^n}{(\tau s + 1)^n} \quad \text{In Series } (k = \tau) \text{ نواحيه.}$$

This can be represented by:

$$\frac{Y(s)}{X(s)} = \frac{k e^{-\tau D}}{\tau s + 1} \quad ; \quad \tau D = n\tau \xrightarrow{\text{Laplace}} \tau = \frac{1}{\nu} \text{ (Time)}$$

$$\frac{Y(s)}{X(s)} = \frac{k^n}{(\tau s + 1)^n} = \frac{k e^{-\tau D}}{\tau s + 1}$$

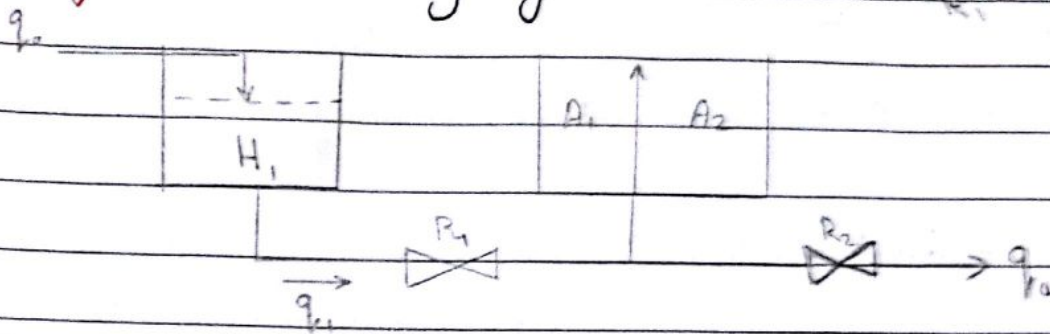
$$\frac{Y(s)}{X(s)} = \frac{k e^{-n\tau s}}{\tau s + 1}$$

It is not identical \*  
 + Series  $\Rightarrow$   
 $\tau = k$

$$\frac{Y(s)}{X(s)} = \frac{k_1 k_2 k_3 \dots}{(\tau_1 s + 1)(\tau_2 s + 1)(\tau_3 s + 1) \dots} \quad \tau \neq k \text{ for all } \tau *$$

Example: Interacting System.

$$Q_1 = \frac{H_1}{R_1}, \quad Q_2 = \frac{H_2}{R_2}$$



find  $G_{ss} = \frac{H_2(s)}{q_i(s)}$

Sol M.B On tank ①

$$q_i - q_o = A_1 \frac{dH_1}{dt}$$

In-Out = ACC ;  $q_i = \frac{H_1 - H_2}{R_1}$   $\leftarrow$  Inter effect acting

$$A \frac{dH_1}{dt} + \frac{H_1 - H_2}{R_1} = q_i$$

$$R_1 A_1 \frac{dH_1}{dt} + H_1 = R_1 q_i + H_2$$

$$\tau_1 \frac{dH_1}{dt} + H_1 = R_1 q_i + H_2$$

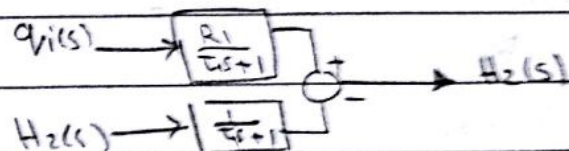
$$(\tau_1 s + 1) H_1(s) = R_1 q_i(s) + H_2(s) \quad \text{①}$$

Transfer Function

M.B On tank ②

$$H_1(s) = \frac{R_1}{\tau_1 s + 1} q_i(s) + \frac{1}{\tau_2 s + 1} H_2(s)$$

$$q_i - q_o = A_2 \frac{dH_2}{dt}$$



$$A_2 \frac{dH_2}{dt} + \frac{H_2}{R_2} = \frac{H_1 - H_2}{R_2}$$

$$A_2 \frac{dH_2}{dt} + \frac{H_2}{R_2} + \frac{H_2}{R_1} = \frac{H_1}{R_1}$$

$$A_2 \frac{dH_2}{dt} + H_2 \left( \frac{R_2 + R_1}{R_1 R_2} \right) = \frac{H_1}{R_1}$$

$$(\tau^* s + 1) H_2(s) = \frac{R_2}{R_1 + R_2} H_1(s) \quad (2)$$

$$(\tau^* s + 1) H_2(s) = K_1 H_1(s)$$

$$\frac{(\tau^* s + 1)(\tau^* s + 1)}{K_1} H_2(s) = R_1 q_v(s) + H_2(s)$$

$$(\tau^* s + 1)(\tau^* s + 1) H_2(s) - K_1 H_2(s) = R_1 K_1 q_v(s)$$

$$(\tau_1 \tau^* s^2 + (\tau_1 + \tau^*) s + 1 - K_1) H_2(s) = R_1 K_1 q_v(s)$$

$$\left( \frac{\tau_1 \tau^*}{1 - K_1} s^2 + \frac{\tau_1 + \tau^*}{1 - K_1} s + 1 \right) H_2(s) = \frac{R_1 K_1}{1 - K_1} q_v(s)$$

$$1 - K_1 = 1 - \frac{R_2}{R_1 + R_2} = \frac{R_1}{R_1 + R_2}$$

$$\frac{\tau^*}{1 - K_1} = \frac{A_2 R_2 R_1 / (R_1 + R_2)}{R_1} = A_2 R_2 = \tau_2$$

$$\frac{\tau_1 + \tau^*}{1 - K_1} = \frac{A_1 R_1 + \frac{A_2 R_1 R_2}{R_1 + R_2}}{R_1} = \frac{A R_1 (R_1 + R_2) + A_2 R_1 R_2}{R_1}$$

$$= A_1 R_1 + A_1 R_2 + A_2 R_2$$

$$= \tau_1 + \tau' + \tau_2$$

$$\frac{R_1 k_1}{1 - k_1} = \frac{\frac{R_2}{R_1 R_2 + R_2}}{\frac{R_1}{R_1 + R_2}} = R_2$$

$$\therefore \frac{H_2(s)}{q_{in}(s)} = \frac{R_2}{T_1 T_2 s^2 + (T_1 + T' + T_1) s}$$

$$T_1 = A_1 R_1 \quad ; \quad T_2 = A_2 R_2$$

$$T' = A_1 R_2$$

## Second Order System

As an example for second order system we can say "Manometer".

after Making a force balance, we get:

$$G(s) = \frac{y(s)}{x(s)} = \frac{K}{\tau^2 s^2 + 2\delta\tau s + 1}$$

where

$T$ : Natural period of oscillation

$$T^2 = \frac{1}{\omega_n^2} \text{ where } \omega_n = \text{natural frequency.}$$

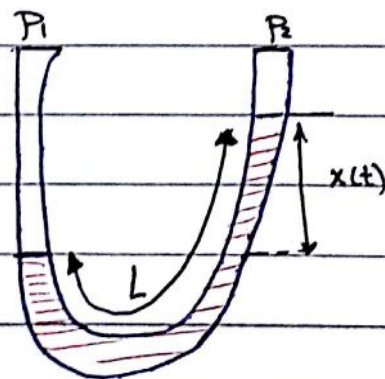
$S$  = damping Factor (coefficient).

We can solve as characteristic eq. like  $(\tau^2 s^2 + 2\delta\tau s + 1 = 0)$

$$s = \frac{-2\delta\tau \pm \sqrt{4\delta^2\tau^2 - 4\tau^2}}{2\tau^2}$$

$$= \frac{2\delta\tau}{2\tau^2} \pm \frac{\sqrt{\delta^2 - 1}}{\tau}$$

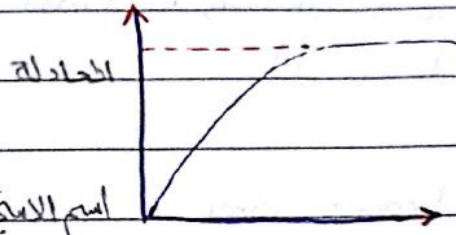
$$s = \frac{-\delta}{\tau} \pm \frac{\sqrt{\delta^2 - 1}}{\tau}$$



Notice that  $\sqrt{\delta^2 - 1}$  determine the nature of response and as follows:

① if  $\delta > 1$  Unequal roots  $(-ve)$  real roots

$$\therefore y = A e^{-m_1 t} + B e^{-m_2 t}$$



Over damped System

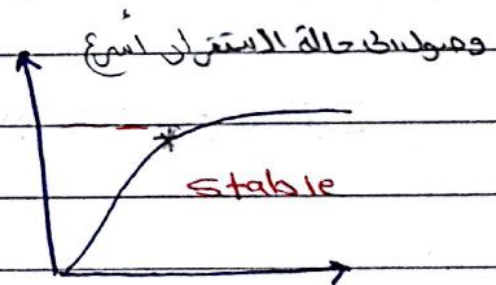
\*not Oscillatory

(غير متذبذب)

\*stable System.

②  $\delta = 1$  equal roots  $(-ve)$ :  $m_1 = m_2$

$$\therefore y = A e^{-mt} + B t e^{-mt} \\ = e^{-mt} (A + B t)$$



Critically damped response. الاستجابة الحرجة

\* Stable and the reaching to the desired Value  
Faster than over damped.

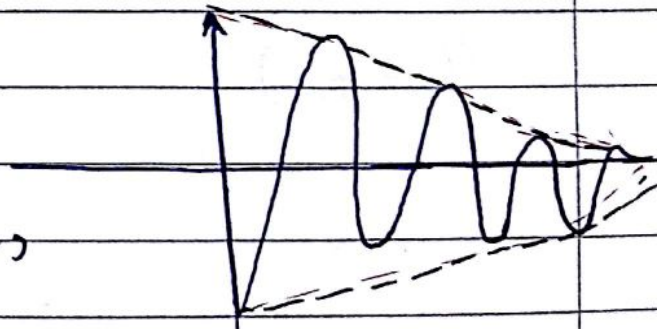
أسرع

③  $0 < \delta < 1 \Rightarrow$  Complex  $(-ve)$  real part

$$s_1 \text{ and } s_2 = -\alpha \pm i\beta$$

$$y = e^{-\alpha t} \sin \beta t$$

Under damped response, stable,  
Oscillation.



\* إن القيمة المثالية لحاصل التناقل للمحول في حالة الاستقرار ستأتي =

$$T = 0.214$$

وأغلب الأحيان في حال عاملها  $\beta$  من النوع الثاني (Under damped).

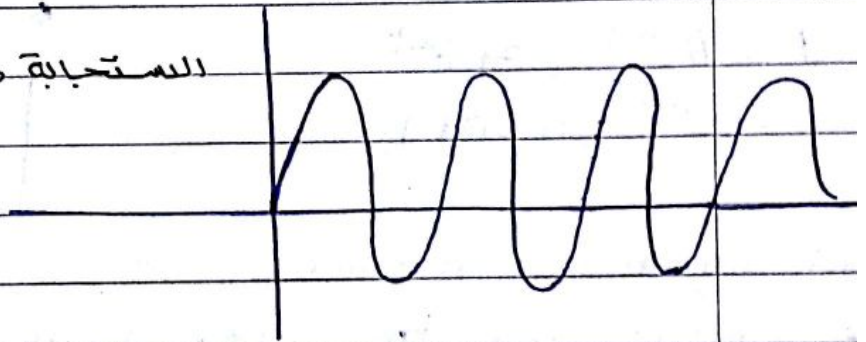
④  $\delta = 0$  pure Imaginary only.

$$s_1 \text{ and } s_2 = \pm i\beta$$

$$y = A \sin \beta t$$

النمط المستقر سعة ثابتة Critically Stable or Free Oscillators.

المستجابة ذات سعة ثابتة

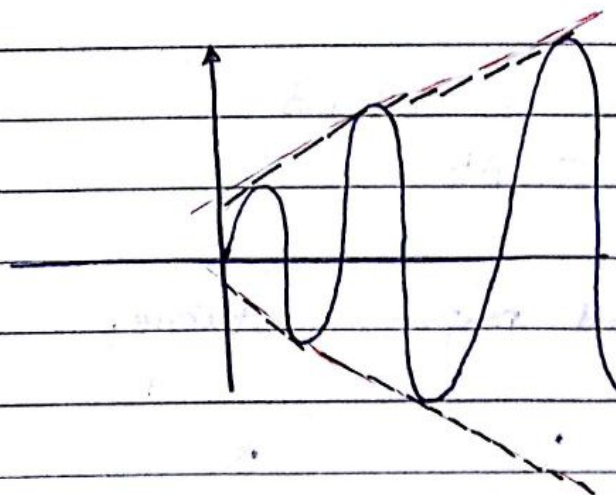


⑤  $\delta < 0$  Complex (+ve real part)

$$s_1 \text{ and } s_2 = +\alpha \pm i\beta$$

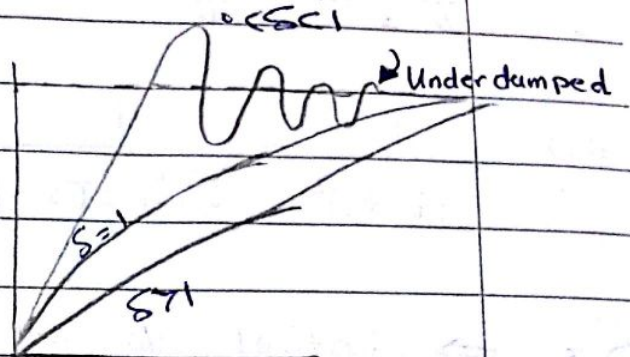
$$y = e^{\alpha t} \sin \beta t$$

(Unstable)

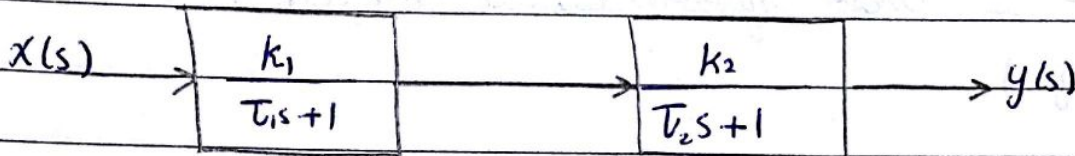


In general for stable system

Optimum  $\delta = 0.214$



Two first Order System In Series



Transfer function :  $\frac{Y(s)}{X(s)} = \frac{k_1 k_2}{(\tau_1 s + 1)(\tau_2 s + 1)} = \frac{k}{\tau_1 \tau_2 s^2 + (\tau_1 + \tau_2)s + 1}$

1  $\tau^2 = \tau_1 \tau_2 \Rightarrow \tau = \sqrt{\tau_1 \tau_2}$   $\tau_1 + \tau_2 = 2\delta\tau$

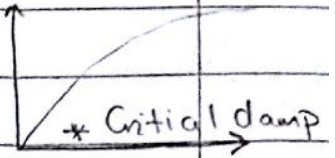
2  $2\delta\tau = \tau_1 + \tau_2 \therefore \tau_1 \neq \tau_2$  (First Order In Series Only)

$\delta = \frac{\tau_1 + \tau_2}{2\tau} = \frac{\tau_1 + \tau_2^2}{2\sqrt{\tau_1 \tau_2}} > 1$  always larger than 1.

∴ Over damped System

If  $\tau_1 = \tau_2 = \tau$

$$\frac{y(s)}{x(s)} = \frac{1}{(\tau s + 1)^2} = \frac{1}{\tau^2 s^2 + 2\tau s + 1} = \frac{1}{\tau^2 s^2 + 2\delta\tau s + 1}$$



$\delta = 1 \Rightarrow$  Critical damped, no oscillation occurs

\* Response of 2<sup>nd</sup> Order System (Under damped to Step Change)

If the Input force is a step change.

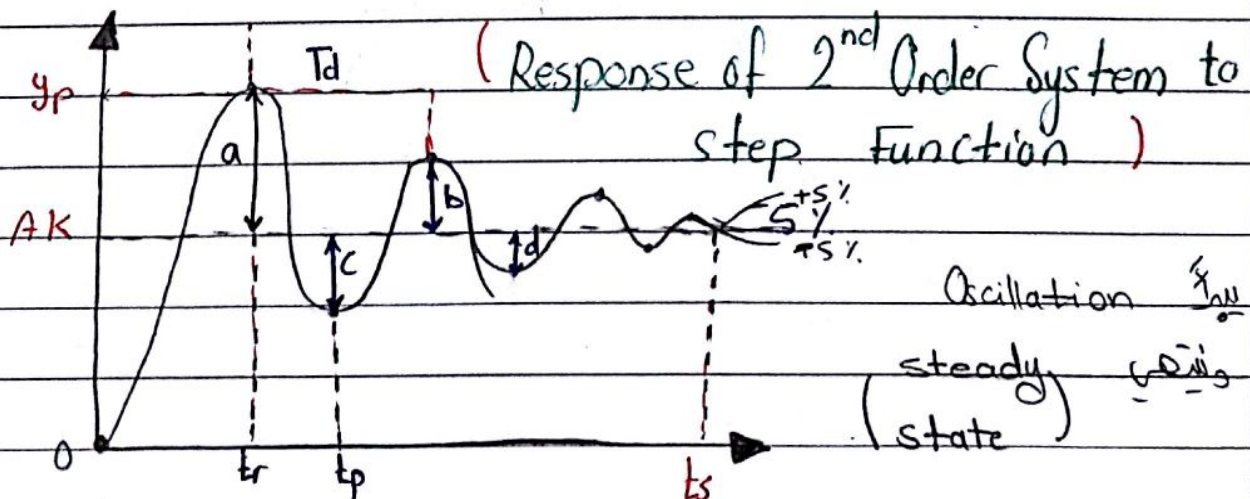
$$y(s) = \frac{K}{s(\tau^2 s^2 + 2\delta\tau s + 1)}$$

Solving by Convolution Laplace المرب \*

$$y(t) = KA \left( 1 - \frac{e^{-st/\tau}}{\sqrt{1-\delta^2}} \sin\left(\frac{\sqrt{1-\delta^2}}{\tau} t + \phi\right) \right)$$

$$\phi = \tan^{-1}\left(\frac{\sqrt{1-\delta^2}}{\delta}\right)$$

\* Force peak القوة \*  
Response Peak  $\theta$



$\omega$ : Angular frequency

$$\omega = \frac{\sqrt{1-\delta^2}}{T} = \omega_n \sqrt{1-\delta^2}$$

Cycle Period:  $T_d$

$$T_d = \frac{2\pi}{\omega} = \frac{2\pi}{\omega_n \sqrt{1-\delta^2}}$$

\* وقت بين peak إلى peak

Time of first Peak:  $t_p$

$$t_p = \frac{\pi T}{\sqrt{1-\delta^2}}$$

\* أول قمة تظهر في الاستجابة

Over shoot

\* هو الفرق بين القيمة الأولى والاستجابة النهائية

\* هو أعلى انحراف عن المواصفات

\* أو القيم المرغوبة (أعلى قيمة)

$$\text{Over shoot} = A_k e^{-\frac{\pi \delta}{\sqrt{1-\delta^2}}}$$

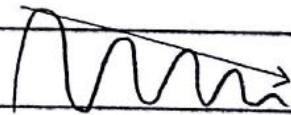
$A_k$  % خروج

$$\% \text{ Over shoot} = \frac{e^{-\frac{\pi \delta}{\sqrt{1-\delta^2}}}}{e} * 100$$

Subsidence Ratio (SR)

نسبة التناؤل

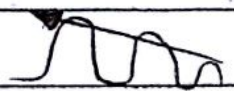
$$SR = \frac{a}{b} = \frac{c}{d} = e^{-\frac{2\pi \delta}{\sqrt{1-\delta^2}}}$$



Decay Ratio (DR)

نسبة التخميد

$$\frac{1}{SR} = \frac{b}{a} = \frac{d}{c} = e^{\frac{2\pi \delta}{\sqrt{1-\delta^2}}} = (DR)$$



$$\text{Optimum DR} = \frac{1}{4} \quad (\text{المطلبي})$$

$$\text{Optimum SR} = 4$$

$$\delta = 0.214$$

Rise time  $t_r$ : This is the time required for the response to reach the first it's Ultimate Value and it's equation:

\* الزمن اللازم لوصول الدالة مرة واحدة

$$t_r = \frac{T(\pi - \phi)}{\sqrt{1 - \delta^2}}$$

$$= \frac{T}{\sqrt{1 - \delta^2}} \left( \pi - \frac{\tan^{-1} \frac{\sqrt{1 - \delta^2}}{\delta}}{\delta} \right)$$

Steady time :  $t_s$  الزمن اللازم لوصول الدالة الى حالة الاستقرار النهائية

والتي يحصل عندما يقل الفرق بين القيمة النهائية الى  $\pm 5\%$

من القيمة المطلوبة

\* استقر ايها عن الزمن عملياً \*

Ex: a/ find SR for the function :

$$f(t) = a e^{-4t} \sin 0.5t$$

b/ find  $\delta$  where  $SR = 4$

Sol/ a-)  $e^{-4t} = e^{-\frac{\delta}{T}t}$

$$4 = \frac{\delta}{T} \quad (1)$$

$$\sin 0.5t = \sin \frac{\sqrt{1-\delta^2}}{T} t$$

$$\frac{\sqrt{1-\delta^2}}{T} = 0.5 \quad (2)$$

From (1) and (2)

$$\frac{\sqrt{1-\delta^2}}{\delta} = \frac{0.4}{5} ; \quad 1-\delta^2 = 0.015625 \delta^2$$

$$\delta^2 = 0.9922$$

$$SR = e^{\frac{2\pi\delta}{\sqrt{1-\delta^2}}} = e^{\frac{2\pi(8)}{16\pi}} = e$$

b/ If  $SR = 4$

$$4 = e^{\frac{2\pi\delta}{\sqrt{1-\delta^2}}}$$

$$1.386 = 2\pi \cdot \frac{\delta}{\sqrt{1-\delta^2}}$$

$$0.22 = \frac{8}{\sqrt{1-\delta^2}} \Rightarrow \delta^* = 0.214^*$$

Response of 2<sup>nd</sup> Order System to Impulse:

$$y(s) = \frac{k}{T^2 s^2 + 2\delta T s + 1} x(s)$$

$$x(s) = A$$

$$y(s) = kA \left( \frac{1}{T^2 s^2 + 2\delta T s + 1} \right)$$

$$y(t) = \frac{kA}{\sqrt{1-\delta^2}T} e^{-\frac{\delta}{T}t} \sin \frac{\sqrt{1-\delta^2}t}{T}$$

bio

Response of 2<sup>nd</sup> Order System for Sinewave.

$$y(s) = \frac{k}{T^2 s^2 + 2\delta T s + 1} x(s)$$

$$x(s) = \frac{aw}{s^2 + \omega^2}, \quad x(t) = a \sin \omega t$$

$$y(t) = \frac{ka}{\sqrt{(1-\omega^2 T^2)^2 + (2\delta \omega T)^2}} \sin(\omega t + \phi)$$

(bio)

$$\phi = \tan^{-1} \frac{2\delta \omega T}{1 - \omega^2 T^2}$$

Ex: The dynamic of particular process are represented by a Single time Constant element in Series with a Second Order (Oscillatory) element. The first Order time Constant is Unity, while the second Order time Constant and damping factor both have a value of 0.2425 min. The steady state gain factor of the process is 0.941. Show that the Overall T.F. Can be Written As:

$$G(s) = \frac{16}{(s+1) [(s+1)^2 + 16]}$$

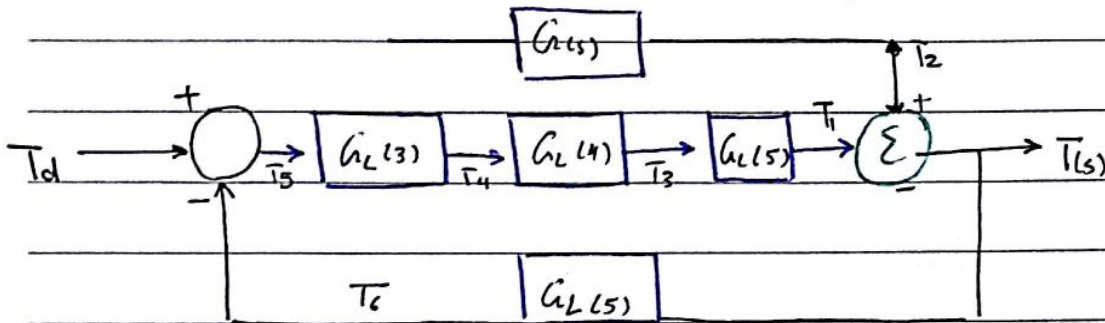
If a Unit Impulse disturbance enter the System, determine the response as a function of time and sketch it.

Sol : Answer  $y(t) = (1 - \cos 4t) e^{-t}$

## Closed System

closed System divided into:

- ① General closed loop system, if he didn't give us any closed loop we can consider it general closed system.



" General closed loop "

Derivation of T.F of general loop:

$$T(s) = T_2 + T_1$$

$$T(s) = m(s) \cdot G_L(s) + T_3 G_P(s)$$

$$T(s) = m(s) \cdot G_L(s) + T_4 G_{LV}(s) \cdot G_P(s)$$

$$T(s) = m(s) \cdot G_L(s) + T_5 G_v \cdot G_c \cdot G_P(s)$$

$$T(s) = m(s) \cdot G_L(s) + (T_d - T_c) G_c \cdot G_P \cdot G_v$$

$$T(s) = m(s) \cdot G_{L(s)} + T_d G_c G_p G_v - T(s) G_m G_c G_v G_p$$

$$T(s) (1 + G_{m(s)} \cdot G_{c(s)} \cdot G_{v(s)} \cdot G_{p(s)}) = m(s) \cdot G_{L(s)} + T_d G_{c(s)} \cdot G_{v(s)} \cdot G_{p(s)}$$

$$\bar{G}_{L(s)} = G_{c(s)} \cdot G_{v(s)} \cdot G_{p(s)}$$

$$T(s) = \frac{G_{L(s)}}{1 + G_{m(s)} \cdot \bar{G}_{L(s)}} m(s) + \frac{\bar{G}_{L(s)}}{1 + G_m \cdot \bar{G}_{L(s)}} T_d$$

Transfer Function of general closed loop can divided into:

### ① Servo System

In which the desired value changing and the load value is constant.

$$G_{L(s)} = \frac{T(s)}{T_d(s)} = \frac{\bar{G}_{L(s)}}{1 + G_{m(s)} + \bar{G}_{L(s)}}$$

### ② Regulator System

In which the desired value is constant and the load value changing

$$G_{L(s)} = \frac{T(s)}{m(s)} = \frac{G_{L(s)}}{1 + G_{m(s)} \cdot \bar{G}_{L(s)}}$$

Ex: Regulator loop with the following elements:

$$K=3, \tau=10 \rightarrow G_p = \frac{3}{10s+1}, \quad G_L(s) = \frac{1}{10s+1}$$

$$G_m(s) = 1, \quad G_v(s) = 1.5, \quad G_c(s) = 2$$

Determine the System response for a unit step in load.

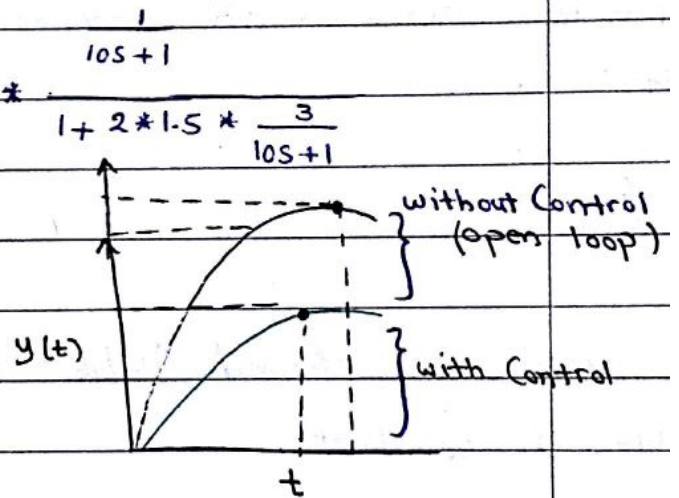
Sol:

$$T(s) = F(s) \cdot G(s)$$

$$= \frac{1}{s} \cdot \frac{G_L(s)}{1 + G_m(s) \cdot G_c(s)} = \frac{1}{s} \cdot \frac{1}{1 + 2 \cdot 1.5 \cdot \frac{3}{10s+1}}$$

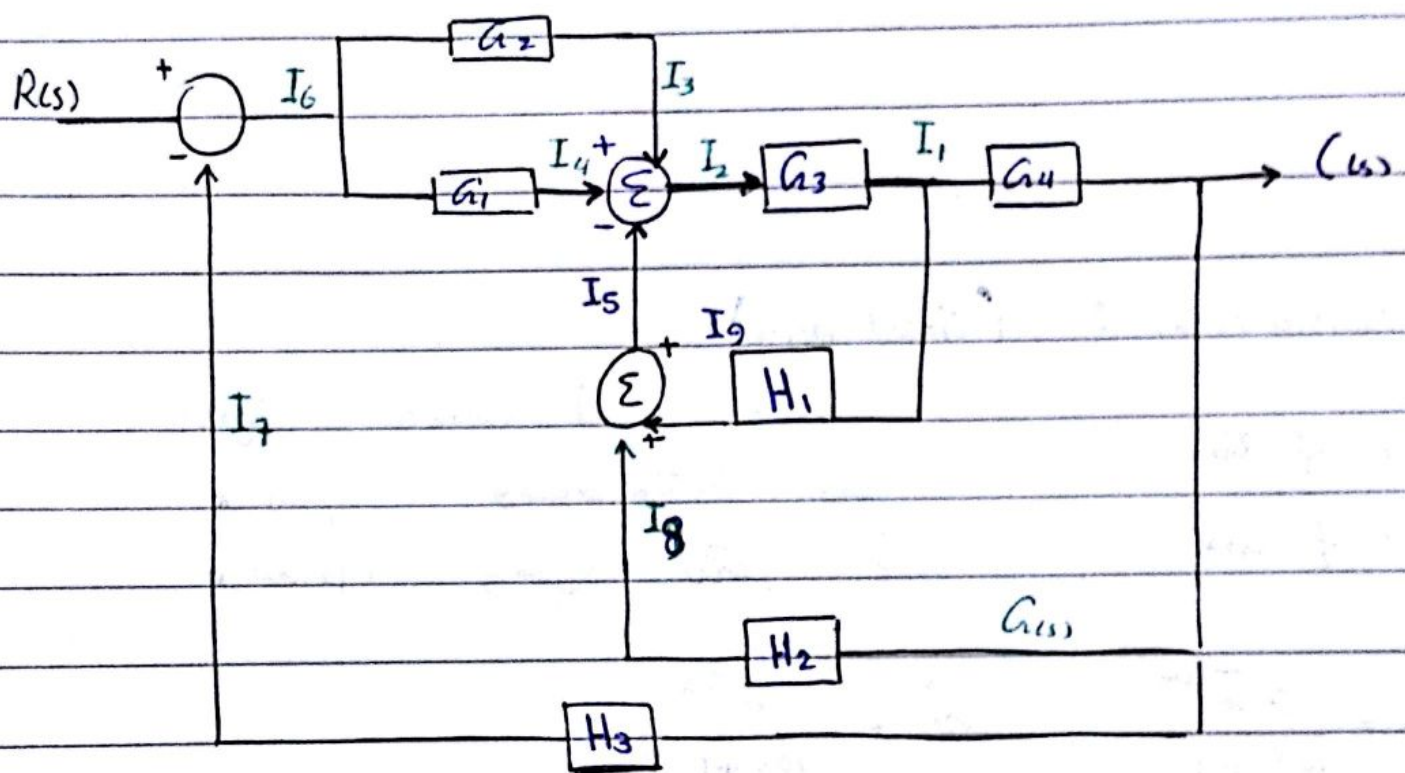
$$= \frac{1}{s} \cdot \frac{\frac{1}{10s+1}}{\frac{10s+1+9}{10s+1}}$$

$$T(s) = \frac{1}{10} (1 - e^{-t})$$



## 2 - Special closed loop System

Ex: Find transfer function  $C(s)/R(s)$  for the System Shown below.



$$C(s) = G_4 I_1$$

$$C(s) = G_4 G_3 I_2 = G_4 G_3 (I_3 + I_4 + I_5)$$

$$C(s) = G_4 G_3 G_2 I_6 + G_4 G_3 G_1 I_6 - G_4 G_3 (I_9 + I_8)$$

$$C(s) = G_4 G_3 G_2 (R(s) - I_7) + G_4 G_3 G_1 (R(s) - I_7) - G_4 G_3 H_1 I_1 - G_4 G_3 H_2 C(s)$$

$$C(s) = G_4 G_3 G_2 R(s) - G_4 G_3 G_2 H_3 C(s) + G_4 G_3 G_1 R(s) - G_4 G_3 G_1 H_3 C(s) - G_4 G_3 H_1 \frac{C(s)}{G_4} - G_4 G_3 H_2 C(s)$$

$$(1 + G_4 G_3 G_2 H_3 + G_4 G_3 G_1 H_3 + G_3 H_1 + G_4 G_3 H_2) C(s)$$

$$= (G_4 G_3 G_2 + G_4 G_3 G_1) R(s)$$

$$\frac{C(s)}{R(s)} = \frac{G_4 G_3 G_2 + G_4 G_3 G_1}{1 + G_4 G_3 G_2 H_3 + G_4 G_3 G_1 H_3 + G_3 H_1 + G_4 G_3 H_2}$$

$$I(s) = I_1 G_4$$

$$I_1 = \frac{C(s)}{G_4(s)}$$

Pade Approximation : (closed loops).

$$e^{-Tos} = \frac{1 - \frac{1}{2} T_0 s}{1 + \frac{1}{2} T_0 s}$$

Open is closed loop

$G_p$  is  $\infty$  : Open

$G_c, G_v$   $\infty$  : Closed

Example:

$$G_p = \frac{3 e^{-Tos}}{10s + 1}, \quad G_L = \frac{e^{-Tos}}{10s + 1}$$

$$G_m(s) = 1, \quad G_v(s) = 1.5, \quad G_d(s) = 2, \quad T_0 = 2$$

Determine the Response For Unit step ; load ?

# Controllers

A Controller is a device that receives data from a measurement instrument, compares that data to a programmed set point, and if necessary, signals a control element to take corrective actions.

↑ Valve

The actions of controllers can be divided into groups based upon the functions of their control mechanism. Each type of controller has advantages and disadvantages and will meet the needs of different applications. Grouped by control mechanism function.

Controllers are:

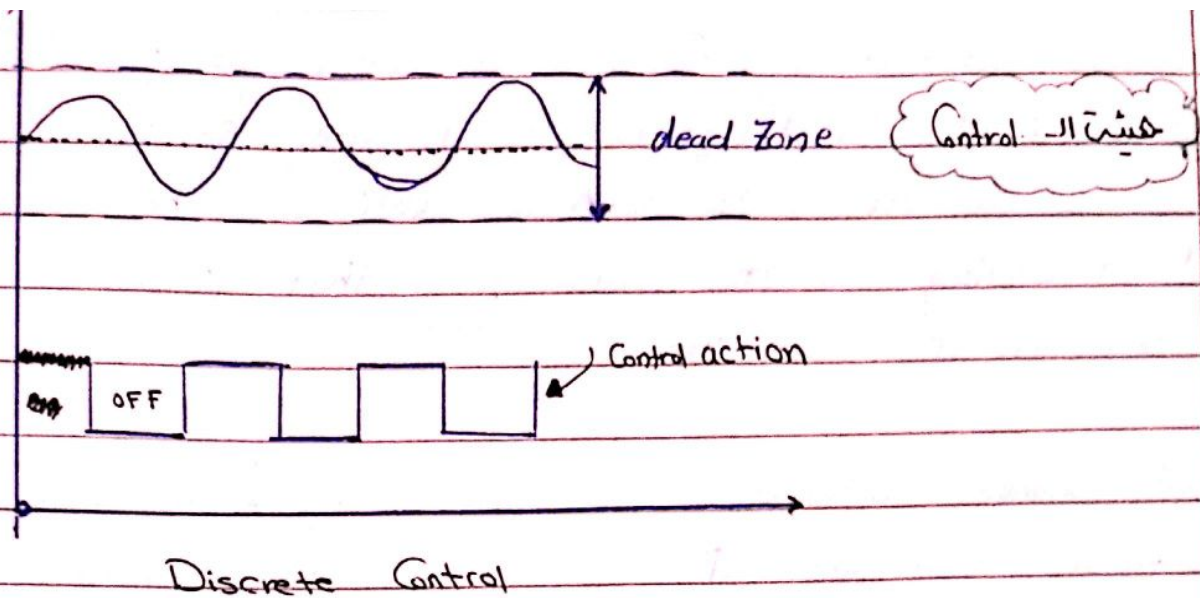
- 1- Discrete Controllers.
- 2- Multi-step Controllers.
- 3- Continuous Controllers.

Discrete Controllers: (On/Off, On/Off, On/Off)

Are controllers that have only two positions: On and Off. A common example of a discrete controller is a home hot water heater.

(Set point بالقياس)

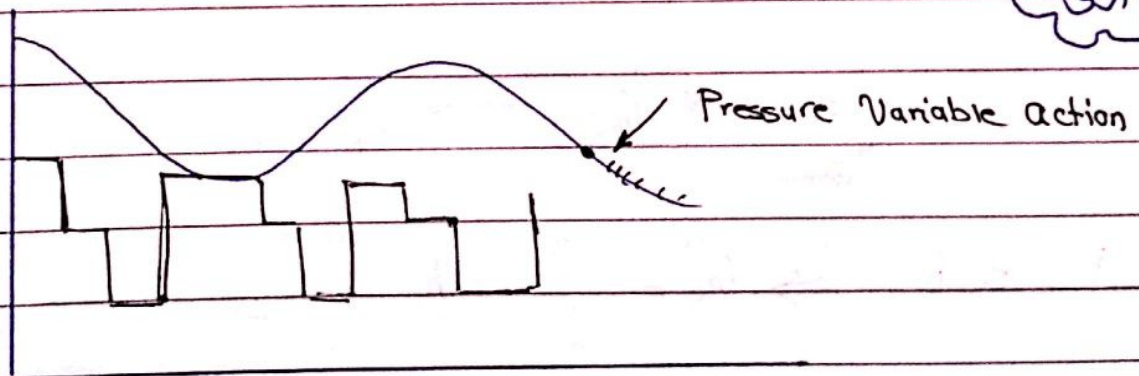
This type of control doesn't actually hold the variable at set point, but keeps the variable within proximity of set point in what is known as a dead zone.



## MultiStep Controllers : (عدة درجات)

Are Controllers that have at least One other possible position in addition to On and off. Multistep Controllers Operate Similarly to discrete Controllers, but as a Setpoint is approached, the multistep Controllers takes intermediate steps.

خطوة وسطيّة



Multistep Controller

هذا تالف خطوات (أفقال) ولأن كل ما كانت الخطوات أكثر فوج

## Continuous Controllers:

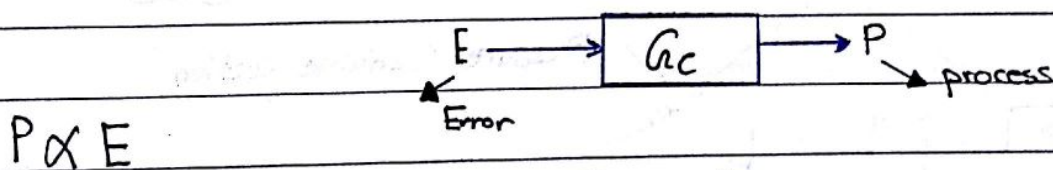
A Controller automatically compares the value of P.V. <sup>Process Variable</sup> to the Set point to determine if an error exists. If there is an error, the Controller <sup>منظم</sup> adjusts the Output according to the parameters <sup>المعيرات</sup> that have been set in the Controller.

### Types of Continuous Controllers:

- ① Proportional Controller (P)
- ② Integral Controller (I)
- ③ Derivation Controller (D)
- ④ Proportional Integral (PI) → <sup>مناظير الـ Controllers المتكاملين</sup>
- ⑤ Proportional Derivate (PD)
- ⑥ Proportional-Integral Derivate (PID) → <sup>معقد المتكاملين</sup>

مقارنة P.V مع S.P  
يوجد error ← خطأ  
لا يوجد error ← سليم

## ① Proportional Controller:

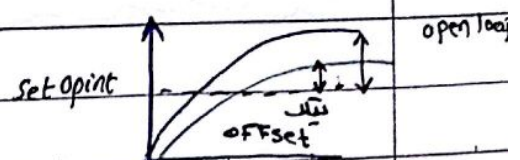


$$P = K_c E \Rightarrow G_c = \frac{P}{E} = K_c$$

$K_c$ : Gain Or Sensitivity of Controller.

$K_c$  ← <sup>مقدار</sup> <sup>تنظيم</sup> <sup>proportional</sup>

- ① Proportional Controller reduce the effect without eliminating it



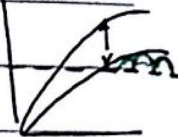


- ② When  $k_c$  Increasing the effect will reduce until we reach  $k_{max}$ , which at it the system starts to regulate for that we use  $k_{opt}$ .

$$k_{optimum} = \frac{k_{c,max}}{2}$$

قليل error  
نقل الحثيث (offset)

أقل حثيث ← تكون قبل التذبذب



- ③ System time constant ( $\tau$ ) is less than process time constant ( $\tau_p$ ) so that the system response will be faster than the process response.

يزيد من الاستجابة  
فيكون مستجيباً عكسياً

- ④ System time constant ( $\tau$ ) decreasing as  $k_c$  increasing

ال error يقل ←  $k_c$  يزداد ←  $\tau$  يقل

## B) Integral Controller :

معدل التذبذب يزداد بزيادة  $k_c$